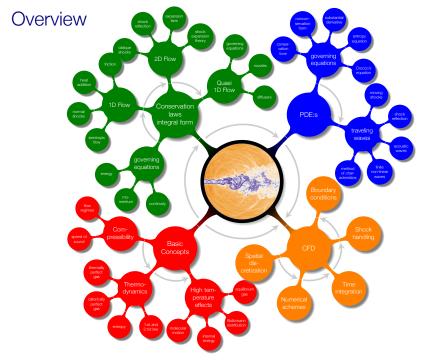
Compressible Flow - TME085 Lecture 15

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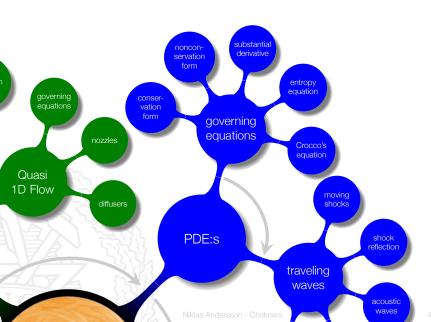
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Chapter 6 Differential Conservation Equations for Inviscid Flows

Overview

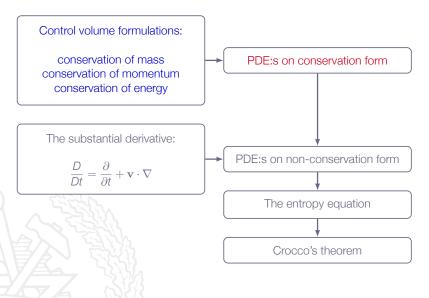


Addressed Learning Outcomes

4 Present at least two different formulations of the governing equations for compressible flows and explain what basic conservation principles they are based on

the governing equations for compressible flows on differential form - finally ...

Roadmap - Differential Equations for Inviscid Flows



Chapter 6.2 Differential Equations in Conservation Form

Differential Equations in Conservation Form

Basic principle to derive PDE:s in conservation form:

- Start with control volume formulation
- Convert to volume integral via Gauss Theorem
- Arbitrary control volume implies that integrand equals to zero everywhere

Continuity Equation

Mass conservation:

Control volume formulation

$$\frac{d}{dt} \iiint_{\Omega} \rho d\mathcal{V} + \iint_{\partial \Omega} \rho \mathbf{v} \cdot \mathbf{n} dS = 0$$

where Ω is a fixed control volume

Applying Gauss Theorem gives

$$\iint_{\partial\Omega} \rho \mathbf{v} \cdot \mathbf{n} dS = \iiint_{\Omega} \nabla \cdot (\rho \mathbf{v}) d\mathcal{V}$$

Also.

$$\frac{\mathrm{d}}{\mathrm{d}t}\iiint\limits_{\Omega}\rho\mathrm{d}\mathscr{V}=\iiint\limits_{\Omega}\frac{\partial\rho}{\partial t}\mathrm{d}\mathscr{V}$$

Continuity Equation

Therefore

$$\iiint\limits_{\Omega} \left[\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) \right] d\mathscr{V} = 0$$

 $\boldsymbol{\Omega}$ is an arbitrary control volume, can be made infinitesimally small and thus

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) = 0$$

which is the continuity equation

Momentum conservation:

Control volume formulation

$$\frac{d}{dt} \iiint_{\Omega} \rho \mathbf{v} d\mathcal{V} + \iint_{\partial \Omega} \left[\rho(\mathbf{v} \cdot \mathbf{n}) \mathbf{v} + \rho \mathbf{n} \right] dS = \iiint_{\Omega} \rho \mathbf{f} d\mathcal{V}$$

where Ω is a fixed control volume

Applying Gauss Theorem gives

$$\iint\limits_{\partial\Omega} \rho(\mathbf{v}\cdot\mathbf{n})\mathbf{v} dS = \iiint\limits_{\Omega} \nabla \cdot (\rho\mathbf{v}\mathbf{v}) d\mathcal{V} \; ; \; \iint\limits_{\partial\Omega} \rho\mathbf{n} dS = \iiint\limits_{\Omega} \nabla \rho d\mathcal{V}$$

Also,

$$\frac{d}{dt} \iiint_{\Omega} \rho \mathbf{v} d\mathcal{V} = \iiint_{\Omega} \frac{\partial}{\partial t} (\rho \mathbf{v}) d\mathcal{V}$$

Therefore

$$\iiint\limits_{\Omega} \left[\frac{\partial}{\partial t} (\rho \mathbf{v}) + \nabla \cdot (\rho \mathbf{v} \mathbf{v}) + \nabla \rho - \rho \mathbf{f} \right] d\mathcal{V} = 0$$

 $\boldsymbol{\Omega}$ is an arbitrary control volume, can be made infinitesimally small and thus

$$\frac{\partial}{\partial t}(\rho \mathbf{v}) + \nabla \cdot (\rho \mathbf{v} \mathbf{v}) + \nabla \rho = \rho \mathbf{f}$$

which is the momentum equation

In cartesian form ($\mathbf{v} = u\mathbf{e}_{x} + v\mathbf{e}_{y} + w\mathbf{e}_{z}$):

$$\frac{\partial}{\partial t}(\rho u) + \nabla \cdot (\rho u \mathbf{v}) + \frac{\partial p}{\partial x} = \rho f_x$$

$$\frac{\partial}{\partial t}(\rho v) + \nabla \cdot (\rho v \mathbf{v}) + \frac{\partial p}{\partial y} = \rho f_y$$

$$\frac{\partial}{\partial t}(\rho w) + \nabla \cdot (\rho w \mathbf{v}) + \frac{\partial p}{\partial z} = \rho f_z$$

or expanded:

$$\frac{\partial}{\partial t}(\rho u) + \frac{\partial}{\partial x}(\rho u u) + \frac{\partial}{\partial y}(\rho u v) + \frac{\partial}{\partial z}(\rho u w) + \frac{\partial p}{\partial x} = \rho f_x$$

$$\frac{\partial}{\partial t}(\rho v) + \frac{\partial}{\partial x}(\rho v u) + \frac{\partial}{\partial y}(\rho v v) + \frac{\partial}{\partial z}(\rho v w) + \frac{\partial p}{\partial y} = \rho f_y$$

$$\frac{\partial}{\partial t}(\rho w) + \frac{\partial}{\partial x}(\rho w u) + \frac{\partial}{\partial y}(\rho w v) + \frac{\partial}{\partial z}(\rho w w) + \frac{\partial p}{\partial z} = \rho f_z$$

$$\frac{\partial}{\partial t}(\rho \mathbf{v}) + \nabla \cdot (\rho \mathbf{v} \mathbf{v}) + \nabla \rho = \rho \mathbf{f}$$

$$\begin{bmatrix} (\rho uu + p) & \rho uv & \rho uw \\ \rho vu & (\rho vv + p) & \rho vw \\ \rho wu & \rho wv & (\rho ww + p) \end{bmatrix} = \rho \mathbf{v}\mathbf{v} + p\mathbf{I}$$

$$\frac{\partial}{\partial t}(\rho \mathbf{v}) + \nabla \cdot (\rho \mathbf{v} \mathbf{v} + \rho \mathbf{I}) = \rho \mathbf{f}$$

Energy Equation

Energy conservation:

Control volume formulation

$$\frac{d}{dt} \iiint_{\Omega} \rho \mathbf{e}_{o} d\mathcal{V} + \iint_{\partial \Omega} \rho h_{o}(\mathbf{v} \cdot \mathbf{n}) dS = \iiint_{\Omega} \rho \mathbf{f} \cdot \mathbf{v} d\mathcal{V}$$

where Ω is a fixed control volume

Applying Gauss Theorem gives

$$\iint_{\partial\Omega} \rho h_o(\mathbf{v} \cdot \mathbf{n}) dS = \iiint_{\Omega} \nabla \cdot (\rho h_o \mathbf{v}) d\mathcal{V}$$

Also,

$$\frac{d}{dt} \iiint \rho \mathbf{e}_0 d\mathscr{V} = \iiint \frac{\partial}{\partial t} (\rho \mathbf{e}_0) d\mathscr{V}$$

Energy Equation

Therefore

$$\iiint\limits_{\Omega} \left[\frac{\partial}{\partial t} (\rho \mathbf{e}_0) + \nabla \cdot (\rho h_0 \mathbf{v}) - \rho (\mathbf{f} \cdot \mathbf{v}) \right] d\mathcal{V} = 0$$

 $\boldsymbol{\Omega}$ is an arbitrary control volume, can be made infinitesimally small and thus

$$\frac{\partial}{\partial t}(\rho \mathbf{e}_{0}) + \nabla \cdot (\rho h_{0} \mathbf{v}) = \rho(\mathbf{f} \cdot \mathbf{v})$$

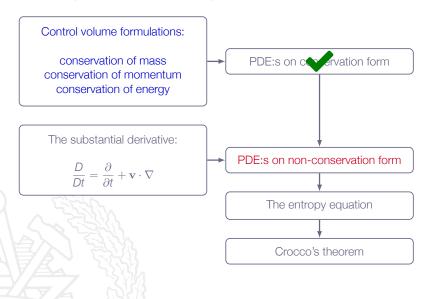
which is the energy equation

Partial Differential Equations in Conservation Form

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) = 0$$
$$\frac{\partial}{\partial t} (\rho \mathbf{v}) + \nabla \cdot (\rho \mathbf{v} \mathbf{v}) + \nabla \rho = \rho \mathbf{f}$$
$$\frac{\partial}{\partial t} (\rho \mathbf{e}_0) + \nabla \cdot (\rho h_0 \mathbf{v}) = \rho (\mathbf{f} \cdot \mathbf{v})$$

These equations are referred to as PDE:s on conservation form since they stem directly from the integral conservation equations applied to a fixed control volume

Roadmap - Differential Equations for Inviscid Flows



Chapter 6.4 Differential Equations in Non-Conservation Form

The Substantial Derivative

Introducing the substantial derivative operator

$$\frac{D}{Dt} = \frac{\partial}{\partial t} + \mathbf{v} \cdot \nabla$$

"... the time rate of change of any quantity associated with a particular moving fluid element is given by the substantial derivative ..."

"... the properties of the fluid element are changing as it moves past a point in a flow because the flowfield itself may be fluctuating with time (the local derivative) and because the fluid element is simply its way to another point in the flowfield where the properties are different (the convective derivative)

Non-Conservation Form of Continuity Equation

Applying the substantial derivative operator to density gives

$$\frac{D\rho}{Dt} = \frac{\partial\rho}{\partial t} + \mathbf{v} \cdot \nabla\rho$$

Continuity equation:

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) = \frac{\partial \rho}{\partial t} + \mathbf{v} \cdot \nabla \rho + \rho (\nabla \cdot \mathbf{v}) = 0 \Rightarrow$$

$$\frac{D\rho}{Dt} + \rho(\nabla \cdot \mathbf{v}) = 0$$

Non-Conservation Form of Continuity Equation

$$\frac{D\rho}{Dt} + \rho(\nabla \cdot \mathbf{v}) = 0$$

"... the mass of a fluid element made up of a fixed set of particles (molecules or atoms) is constant as the fluid element moves through space ..."

Non-Conservation Form of Momentum Equation

$$\frac{\partial}{\partial t}(\rho \mathbf{v}) + \nabla \cdot (\rho \mathbf{v} \mathbf{v} + \rho \mathbf{I}) = \rho \mathbf{f} \Rightarrow$$

$$\rho \frac{\partial \mathbf{v}}{\partial t} + \mathbf{v} \frac{\partial \rho}{\partial t} + \rho \mathbf{v} \cdot \nabla \mathbf{v} + \mathbf{v}(\nabla \cdot \rho \mathbf{v}) + \nabla \rho = \rho \mathbf{f} \Rightarrow$$

$$\rho \underbrace{\left[\frac{\partial \mathbf{v}}{\partial t} + \mathbf{v} \cdot \nabla \mathbf{v}\right]}_{=\overline{Dt}} + \mathbf{v} \underbrace{\left[\frac{\partial \rho}{\partial t} + \nabla \cdot \rho \mathbf{v}\right]}_{=0} + \nabla \rho = \rho \mathbf{f}$$

$$\frac{D\mathbf{v}}{Dt} + \frac{1}{\rho} \nabla \rho = \mathbf{f}$$

$$\frac{\partial}{\partial t}(\rho \mathbf{e}_{o}) + \nabla \cdot (\rho h_{o} \mathbf{v}) = \rho(\mathbf{f} \cdot \mathbf{v}) + \rho \dot{q}$$

$$h_{o} = \mathbf{e}_{o} + \frac{\rho}{\rho} \Rightarrow$$

$$\frac{\partial}{\partial t}(\rho \mathbf{e}_{o}) + \nabla \cdot (\rho \mathbf{e}_{o} \mathbf{v}) + \nabla \cdot (\rho \mathbf{v}) = \rho(\mathbf{f} \cdot \mathbf{v}) + \rho \dot{q} \Rightarrow$$

$$\rho \frac{\partial \mathbf{e}_{o}}{\partial t} + \mathbf{e}_{o} \frac{\partial \rho}{\partial t} + \rho \mathbf{v} \cdot \nabla \mathbf{e}_{o} + \mathbf{e}_{o} \nabla \cdot (\rho \mathbf{v}) + \nabla \cdot (\rho \mathbf{v}) = \rho(\mathbf{f} \cdot \mathbf{v}) + \rho \dot{q} \Rightarrow$$

$$\rho \underbrace{\left[\frac{\partial \mathbf{e}_o}{\partial t} + \mathbf{v} \cdot \nabla \mathbf{e}_o\right]}_{=\frac{D\mathbf{e}_o}{\partial t}} + \mathbf{e}_o \underbrace{\left[\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v})\right]}_{=0} + \nabla \cdot (\rho \mathbf{v}) = \rho(\mathbf{f} \cdot \mathbf{v}) + \rho \dot{q}$$

$$\rho \frac{De_o}{Dt} + \nabla \cdot (\rho + \mathbf{v}) = \rho \mathbf{f} \cdot \mathbf{v} + \rho \dot{q}$$
$$e_o = e + \frac{1}{2} \mathbf{v} \cdot \mathbf{v} \Rightarrow$$

$$\rho \frac{D\mathbf{e}}{Dt} + \rho \mathbf{v} \cdot \frac{D\mathbf{v}}{Dt} + \nabla \cdot (\rho \mathbf{v}) = \rho \mathbf{f} \cdot \mathbf{v} + \rho \dot{q}$$

Using the momentum equation, $\left(\frac{D\mathbf{v}}{Dt} + \frac{1}{\rho}\nabla p = \mathbf{f}\right)$, gives

$$\rho \frac{De}{Dt} - \mathbf{v} \cdot \nabla \rho + \rho \mathbf{f} \cdot \mathbf{v} + \mathbf{v} \cdot \nabla \rho + \rho (\nabla \cdot \mathbf{v}) = \rho \mathbf{f} \cdot \mathbf{v} + \rho \dot{q} \Rightarrow$$

$$\frac{D\mathbf{e}}{Dt} + \frac{p}{\rho}(\nabla \cdot \mathbf{v}) = \dot{q}$$

$$\frac{De}{Dt} + \frac{p}{\rho}(\nabla \cdot \mathbf{v}) = \dot{q}$$

From the continuity equation we get

$$\frac{D\rho}{Dt} + \rho(\nabla \cdot \mathbf{v}) = 0 \Rightarrow \nabla \cdot \mathbf{v} = -\frac{1}{\rho} \frac{D\rho}{Dt} \Rightarrow$$

$$\frac{De}{Dt} - \frac{\rho}{\rho^2} \frac{D\rho}{Dt} = \dot{q} \Rightarrow \frac{De}{Dt} + \rho \frac{D}{Dt} \left(\frac{1}{\rho}\right) = \dot{q}$$

$$\left[\begin{array}{c} \frac{De}{Dt} = \dot{q} - \rho \frac{D\nu}{Dt} \end{array}\right]$$

where $\nu = 1/\rho$

Compare with first law of thermodynamics: $de = \delta q - \delta W$

If we instead express the energy equation in terms of enthalpy:

$$\frac{De}{Dt} = \dot{q} - \rho \frac{D}{Dt} \left(\frac{1}{\rho} \right) \Rightarrow \frac{De}{Dt} + \rho \frac{D}{Dt} \left(\frac{1}{\rho} \right) = \dot{q}$$

$$h = e + \frac{p}{\rho} \Rightarrow \frac{Dh}{Dt} = \frac{De}{Dt} + \frac{1}{\rho} \frac{Dp}{Dt} + p \frac{D}{Dt} \left(\frac{1}{\rho}\right) \Rightarrow$$

$$\frac{Dh}{Dt} = \dot{q} + \frac{1}{\rho} \frac{Dp}{Dt}$$

and total enthalpy ...

$$h_0 = h + \frac{1}{2} \mathbf{v} \cdot \mathbf{v} \Rightarrow \frac{Dh_0}{Dt} = \frac{Dh}{Dt} + \mathbf{v} \cdot \frac{D\mathbf{v}}{Dt}$$

From the momentum equation we get

$$\rho \frac{\mathbf{D}\mathbf{v}}{\mathbf{D}t} + \nabla \rho = \rho \mathbf{f} \Rightarrow \frac{\mathbf{D}\mathbf{v}}{\mathbf{D}t} = -\frac{1}{\rho} \nabla \rho + \mathbf{f} \Rightarrow$$

$$\frac{Dh_o}{Dt} = \underbrace{\frac{Dh}{Dt}}_{\dot{q} + \frac{1}{\rho} \frac{Dp}{Dt}} - \frac{1}{\rho} \mathbf{v} \cdot \nabla p + \mathbf{f} \cdot \mathbf{v} \Rightarrow$$

$$\frac{Dh_o}{Dt} = \dot{q} + \frac{1}{\rho} \left[\frac{Dp}{Dt} - \mathbf{v} \cdot \nabla p \right] + \mathbf{f} \cdot \mathbf{v}$$

$$\frac{Dh_o}{Dt} = \dot{q} + \frac{1}{\rho} \left[\frac{D\rho}{Dt} - \mathbf{v} \cdot \nabla \rho \right] + \mathbf{f} \cdot \mathbf{v}$$

Expanding the substantial derivative $\frac{Dp}{Dt}$ gives

$$\frac{D\rho}{Dt} = \frac{\partial \rho}{\partial t} + \mathbf{v} \cdot \nabla \rho \Rightarrow$$

$$\frac{Dh_o}{Dt} = \frac{1}{\rho} \frac{\partial p}{\partial t} + \dot{q} + \mathbf{f} \cdot \mathbf{v}$$

Let's examine the above relation ...

$$\frac{Dh_o}{Dt} = \frac{1}{\rho} \frac{\partial p}{\partial t} + \dot{q} + \mathbf{f} \cdot \mathbf{v}$$

The total enthalpy of a moving fluid element in an inviscid flow can change due to

- unsteady flow: $\partial p/\partial t \neq 0$
- heat transfer: $\dot{q} \neq 0$
- body forces: $\mathbf{f} \cdot \mathbf{v} \neq 0$

Adiabatic flow without body forces ⇒

$$\frac{Dh_o}{Dt} = \frac{1}{\rho} \frac{\partial p}{\partial t}$$

Steady-state adiabatic flow without body forces \Rightarrow

$$\frac{Dh_o}{Dt} = 0$$

 h_0 is constant along streamlines!

Start from

$$\frac{De}{Dt} = \dot{q} - \rho \frac{D}{Dt} \left(\frac{1}{\rho} \right)$$

Calorically perfect gas:

$$e = C_v T$$
; $C_v = \frac{R}{\gamma - 1}$; $p = \rho RT$; $\gamma, R = const$

$$\frac{De}{Dt} = C_V \frac{DT}{Dt} = \frac{R}{\gamma - 1} \frac{D}{Dt} \left(\frac{\rho}{\rho R} \right) = \frac{1}{\gamma - 1} \frac{D}{Dt} \left(\frac{\rho}{\rho} \right) \Rightarrow$$

$$rac{1}{\gamma-1}rac{D}{Dt}\left(rac{
ho}{
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ight)=\dot{q}-
horac{D}{Dt}\left(rac{1}{
ho}
ight)\Rightarrow$$

$$\frac{1}{\gamma - 1} \left[\rho \frac{D}{Dt} \left(\frac{1}{\rho} \right) + \left(\frac{1}{\rho} \right) \frac{D\rho}{Dt} \right] = \dot{q} - \rho \frac{D}{Dt} \left(\frac{1}{\rho} \right)$$

$$\rho \frac{D}{Dt} \left(\frac{1}{\rho} \right) + \left(\frac{1}{\rho} \right) \frac{D\rho}{Dt} = (\gamma - 1) \dot{q} - (\gamma - 1) \rho \frac{D}{Dt} \left(\frac{1}{\rho} \right)$$

$$\gamma p \frac{D}{Dt} \left(\frac{1}{\rho} \right) + \left(\frac{1}{\rho} \right) \frac{Dp}{Dt} = (\gamma - 1)\dot{q}$$

Continuity:

$$\frac{D\rho}{Dt} = -\rho(\nabla \cdot \mathbf{v}) \Rightarrow \frac{D}{Dt} \left(\frac{1}{\rho}\right) = -\frac{1}{\rho^2} \frac{D\rho}{Dt} = \frac{1}{\rho} (\nabla \cdot \mathbf{v}) \Rightarrow$$

$$\frac{\gamma \rho}{\rho} (\nabla \cdot \mathbf{v}) + \left(\frac{1}{\rho}\right) \frac{D\rho}{Dt} = (\gamma - 1)\dot{q}$$

$$\frac{Dp}{Dt} + \gamma p(\nabla \cdot \mathbf{v}) = (\gamma - 1)\rho \dot{q}$$

Adiabatic flow (no added heat):

$$\frac{D\rho}{Dt} + \gamma \rho(\nabla \cdot \mathbf{v}) = 0$$

Non-conservation form (calorically perfect gas)

Conservation Form

$$\frac{\partial Q}{\partial t} + \frac{\partial E}{\partial x} + \frac{\partial F}{\partial y} + \frac{\partial G}{\partial z} = 0$$

where Q(x,y,z,t), E(x,y,z,t), ... may be scalar or vector fields

Example: the continuity equation

$$\frac{\partial \rho}{\partial t} + \frac{\partial}{\partial x}(\rho u) + \frac{\partial}{\partial y}(\rho v) + \frac{\partial}{\partial z}(\rho w) = 0$$

If an equation cannot be written in this form, it is said to be in non-conservation form

Euler Equations - Conservation Form

Continuity, momentum and energy equations in Cartesian coordinates, velocity components u, v, w (no body forces, no added heat)

$$\frac{\partial \rho}{\partial t} + \frac{\partial}{\partial x}(\rho u) + \frac{\partial}{\partial y}(\rho v) + \frac{\partial}{\partial z}(\rho w) = 0$$

$$\frac{\partial}{\partial t}(\rho u) + \frac{\partial}{\partial x}(\rho u u + \rho) + \frac{\partial}{\partial y}(\rho u v) + \frac{\partial}{\partial z}(\rho u w) = 0$$

$$\frac{\partial}{\partial t}(\rho v) + \frac{\partial}{\partial x}(\rho v u) + \frac{\partial}{\partial y}(\rho v v + \rho) + \frac{\partial}{\partial z}(\rho v w) = 0$$

$$\frac{\partial}{\partial t}(\rho w) + \frac{\partial}{\partial x}(\rho w u) + \frac{\partial}{\partial y}(\rho w v) + \frac{\partial}{\partial z}(\rho w w + \rho) = 0$$

$$\frac{\partial}{\partial t}(\rho e_o) + \frac{\partial}{\partial x}(\rho h_o u) + \frac{\partial}{\partial y}(\rho h_o v) + \frac{\partial}{\partial z}(\rho h_o w) = 0$$

Euler Equations - Non-Conservation Form

Continuity, momentum and energy equations in Cartesian coordinates, velocity components u, v, w (no body forces, no added heat), calorically perfect gas

$$\frac{\partial \rho}{\partial t} + u \frac{\partial \rho}{\partial x} + v \frac{\partial \rho}{\partial y} + w \frac{\partial \rho}{\partial z} + \rho \left(\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} \right) = 0$$

$$\frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} + w \frac{\partial u}{\partial z} + \frac{1}{\rho} \frac{\partial \rho}{\partial x} = 0$$

$$\frac{\partial v}{\partial t} + u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} + w \frac{\partial v}{\partial z} + \frac{1}{\rho} \frac{\partial \rho}{\partial y} = 0$$

$$\frac{\partial w}{\partial t} + u \frac{\partial w}{\partial x} + v \frac{\partial w}{\partial y} + w \frac{\partial w}{\partial z} + \frac{1}{\rho} \frac{\partial \rho}{\partial z} = 0$$

$$\frac{\partial \rho}{\partial t} + u \frac{\partial \rho}{\partial x} + v \frac{\partial \rho}{\partial y} + w \frac{\partial \rho}{\partial z} + \gamma \rho \left(\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} \right) = 0$$

Conservation and Non-Conservation Form

The governing equations on non-conservation form are not, although the name might give that impression, less physically accurate than the equations on conservation form. The nomenclature comes from CFD where the equations on conservation form are preferred.

Conservation and Non-Conservation Form

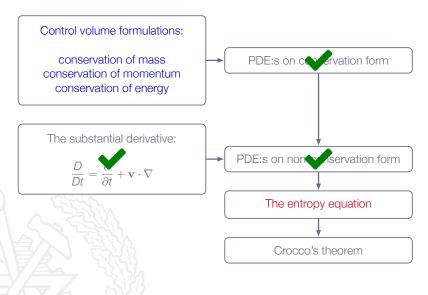
Conservation forms are useful for:

- 1. Numerical methods for compressible flow
- 2. Theoretical understanding of non-linear waves (shocks etc)
- 3. Provide link between integral forms (control volume formulations) and PDE:s

Non-conservation forms are useful for:

- Theoretical understanding of behavior of numerical methods
- 2. Theoretical understanding of boundary conditions
- 3. Analysis of linear waves (aero-acoustics)

Roadmap - Differential Equations for Inviscid Flows



Chapter 6.5 The Entropy Equation

The Entropy Equation

From the first and second law of thermodynamics we have

$$\frac{De}{Dt} = T\frac{Ds}{Dt} - \rho \frac{D}{Dt} \left(\frac{1}{\rho}\right)$$

which is called the entropy equation

The Entropy Equation

Compare the entropy equation

$$\frac{De}{Dt} = T\frac{Ds}{Dt} - \rho \frac{D}{Dt} \left(\frac{1}{\rho}\right)$$

with the energy equation (inviscid flow):

$$\frac{De}{Dt} = \dot{q} - \rho \frac{D}{Dt} \left(\frac{1}{\rho} \right)$$

we see that

$$T\frac{Ds}{Dt} = \dot{q}$$

The Entropy Equation

If $\dot{q} = 0$ (adiabatic flow) then

$$\frac{Ds}{Dt} = 0$$

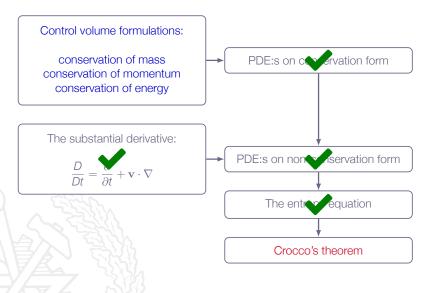
i.e., entropy is constant for moving fluid element

Furthermore, if the flow is steady we have

$$\frac{Ds}{Dt} = \frac{\partial s}{\partial t} + (\mathbf{v} \cdot \nabla)s = (\mathbf{v} \cdot \nabla)s = 0$$

i.e., entropy is constant along streamlines

Roadmap - Differential Equations for Inviscid Flows



Chapter 6.6 Crocco's Theorem



"... a relation between gradients of total enthalpy, gradients of entropy, and flow rotation ..."



Momentum equation (no body forces)

$$\rho \frac{D\mathbf{v}}{Dt} = -\nabla \rho$$

Writing out the substantial derivative gives

$$\rho \frac{\partial \mathbf{v}}{\partial t} + \rho \mathbf{v} \cdot \nabla \mathbf{v} = -\nabla \rho \Rightarrow \frac{\partial \mathbf{v}}{\partial t} + \mathbf{v} \cdot \nabla \mathbf{v} = -\frac{1}{\rho} \nabla \rho$$

First and second law of thermodynamics (energy equation)

$$dh = Tds + \frac{1}{\rho}dp$$

Replace differentials with a gradient operator

$$\nabla h = T \nabla s + \frac{1}{\rho} \nabla p \Rightarrow T \nabla s = \nabla h - \frac{1}{\rho} \nabla p$$

With pressure derivative from the momentum equation inserted in the energy equation we get

$$T\nabla s = \nabla h + \frac{\partial \mathbf{v}}{\partial t} + \mathbf{v} \cdot \nabla \mathbf{v}$$

$$h = h_0 - \frac{1}{2} \mathbf{v} \cdot \mathbf{v} \Rightarrow \nabla h = \nabla h_0 - \nabla (\frac{1}{2} \mathbf{v} \cdot \mathbf{v})$$

$$\nabla(\frac{1}{2}\mathbf{v}\cdot\mathbf{v}) = \mathbf{v}\times(\nabla\times\mathbf{v}) + \mathbf{v}\cdot\nabla\mathbf{v}$$

$$\nabla (\mathbf{A} \cdot \mathbf{B}) = (\mathbf{A} \cdot \nabla) \mathbf{B} + (\mathbf{B} \cdot \nabla) \mathbf{A} + \mathbf{A} \times (\nabla \times \mathbf{B}) + \mathbf{B} \times (\nabla \times \mathbf{A})$$

$$A = B = v \Rightarrow$$

$$\nabla(\mathbf{v} \cdot \mathbf{v}) = 2[\mathbf{v} \cdot \nabla \mathbf{v} + \mathbf{v} \times (\nabla \times \mathbf{v})]$$

$$T\nabla s = \nabla h_o - \mathbf{v} \times (\nabla \times \mathbf{v}) - \mathbf{v} \cdot \nabla \mathbf{v} + \frac{\partial \mathbf{v}}{\partial t} + \mathbf{v} \cdot \nabla \mathbf{v}$$

$$T\nabla s = \nabla h_0 + \frac{\partial \mathbf{v}}{\partial t} - \mathbf{v} \times (\nabla \times \mathbf{v})$$

Note: $\nabla \times \mathbf{v}$ is the vorticity of the fluid

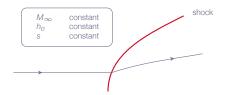
the rotational motion of the fluid is described by the angular velocity $\omega = \frac{1}{2}(\nabla \times \mathbf{v})$

$$T\nabla s = \nabla h_0 + \frac{\partial \mathbf{v}}{\partial t} - \mathbf{v} \times (\nabla \times \mathbf{v})$$

"... when a steady flow field has gradients of total enthalpy and/or entropy Crocco's theorem dramatically shows that it is rotational ..."

Crocco's Theorem - Example

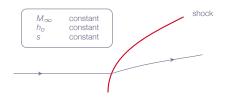
Curved stationary shock (steady-state flow)



- s is constant upstream of shock
- jump in s across shock depends on local shock angle
- s will vary from streamline to streamline downstream of shock
- $\nabla s \neq 0$ downstream of shock

Crocco's Theorem - Example

Curved stationary shock (steady-state flow)



- ► Total enthalpy upstream of shock
 - ho is constant along streamlines
 - ▶ h₀ is uniform
- Total enthalpy downstream of shock
 - \triangleright h_0 is uniform

$$\nabla h_0 = 0$$

Crocco's Theorem - Example

Crocco's equation for steady-state flow:

$$T\nabla s = \nabla h_o - \mathbf{v} \times (\nabla \times \mathbf{v})$$

- $\mathbf{v} \times (\nabla \times \mathbf{v}) \neq 0$ downstream of a curved shock
- ▶ the rotation $\nabla \times \mathbf{v} \neq 0$ downstream of a curved shock

Explains why it is difficult to solve such problems by analytic means!

Roadmap - Differential Equations for Inviscid Flows

