

# DETACHED EDDY SIMULATION COUPLED WITH STEADY RANS IN THE WALL REGION

Lars Davidson

Division of Fluid Dynamics  
Dept. of Mechanics and Maritime Sciences  
SE-412 96 Gothenburg, Sweden

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- ▶ Solution:
  - ▶ solve the steady equations in the RANS region

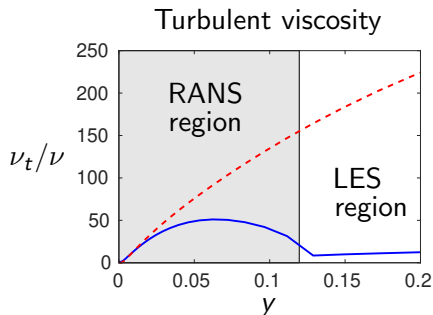


FIGURE: Channel flow. — : DES; - - ; 1D steady RANS.

# TWO SOLVERS

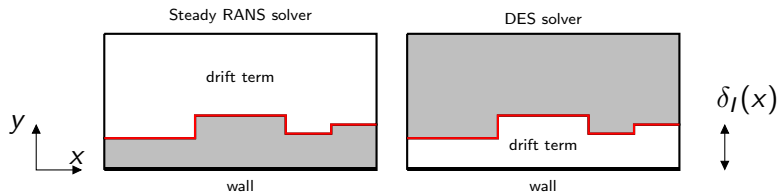


FIGURE: Grey color indicates the solver that drives the flow

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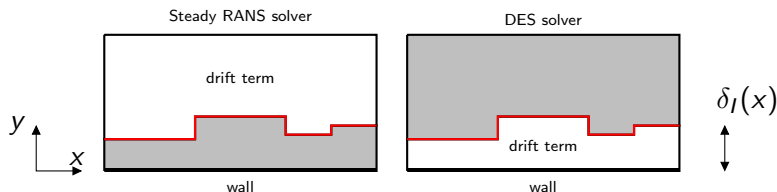


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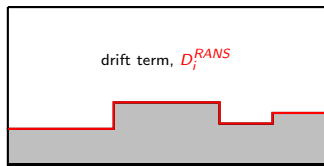
- ▶ The interface,  $\delta_I(x)$ , is defined by the usual DES switch

$$F_{DES} = \max \left\{ \frac{L_t}{\Delta}, 1 \right\} = \max \left\{ \frac{k^{1/2}/(C_\mu \omega)}{\Delta}, 1 \right\}$$



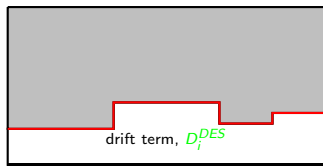
# MOMENTUM EQUATIONS

Steady RANS solver



wall

DES solver



wall

$$\frac{\partial \langle \bar{v}_j \rangle \langle \bar{v}_i \rangle}{\partial x_j} = S_i^{RANS} - \frac{1}{\rho} \frac{\partial \langle \bar{p} \rangle}{\partial x_i}$$

$$+ \frac{\partial}{\partial x_j} \left( (\nu + \langle \nu_t \rangle) \frac{\partial \langle \bar{v}_i \rangle}{\partial x_j} \right)$$

$$\langle \bar{v}_i \rangle^{RANS} \Rightarrow \langle \bar{v}_i \rangle_T^{DES}$$

$$S_i^{RANS} = \frac{\langle v_i^{DES} \rangle_T - \langle v_i^{RANS} \rangle}{10^{-10}}$$

$$\frac{\partial \bar{v}_i}{\partial t} + \frac{\partial \bar{v}_j \bar{v}_i}{\partial x_j} = S_i^{DES} - \frac{1}{\rho} \frac{\partial \bar{p}}{\partial x_i}$$

$$+ \frac{\partial}{\partial x_j} \left( (\nu + \nu_t) \frac{\partial \bar{v}_i}{\partial x_j} \right)$$

$$\langle \bar{v}_i \rangle_T^{DES} \Rightarrow \langle \bar{v}_i \rangle^{RANS}$$

$$S_i^{DES} = \frac{\langle v_i^{RANS} \rangle - \langle \bar{v}_i^{DES} \rangle_T}{\tau_r}$$

# TIME AVERAGING

- ▶ Subscript  $T$  indicates averaging time (memory =  $T$ )

$$\langle \phi(t) \rangle_T = \frac{1}{T} \int_{-\infty}^t \phi(\tau) \exp(-(t - \tau)/T) d\tau \Rightarrow$$

$$\langle \phi \rangle_T^t \equiv \langle \phi \rangle_T = a \langle \phi \rangle_T^{t - \Delta t} + (1 - a) \phi^t$$

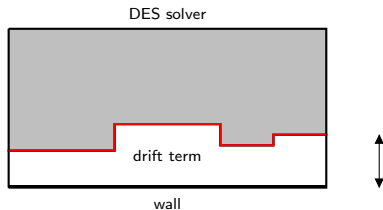
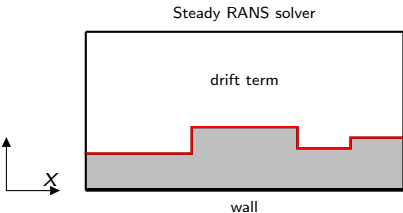
$$a = \exp(-\Delta t/T).$$

$$\tau_r = \max(0.1k/\varepsilon, \Delta t)$$

# NUMERICAL METHOD: FINITE VOLUME METHODS

- ▶ CALC-BFC [6]
- ▶ 2D RANS solver
- ▶ Staggered grid
- ▶ SIMPLEC
- ▶  $\bar{u}$ ,  $\bar{v}$ ,  $\bar{w}$ : 2nd order upwind
- ▶  $k$  &  $\omega$ : Hybrid 1<sup>st</sup> order
- ▶ Steady
- ▶ CALC-LES [5]
- ▶ 3D DES solver
- ▶ Collocated grid
- ▶ Pressure-velocity: fractional step
- ▶  $\bar{u}$ ,  $\bar{v}$ ,  $\bar{w}$ : Central differences (CDS)
- ▶  $k$  &  $\omega$ : Hybrid 1<sup>st</sup> order upwind/C
- ▶ Crank-Nicholson in time

# TURBULENCE MODELS



- ▶ EARSM (Explicit Algebraic Stress Model) [9]
- ▶ coupled to Wilcox  $k - \omega$  model [10]

- ▶ DES  $k - \omega$  model

# COUPLING IN THE RANS SOLVER

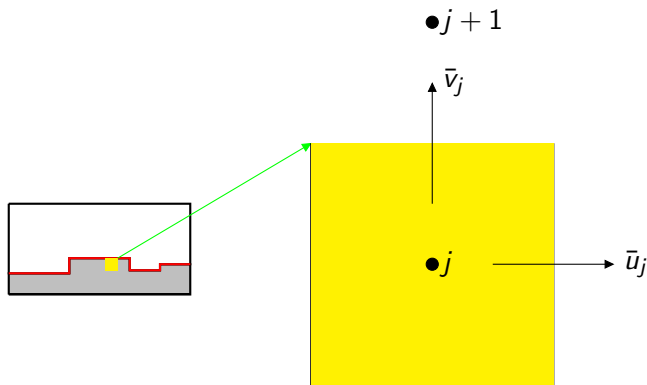


FIGURE: Cell  $j$  in RANS solver adjacent to the interface.

- ▶  $\langle \bar{p}_{j+1}^{LES} \rangle_T$ ,  $\langle \bar{u}_{j+1}^{LES} \rangle_T$ , at the LES-RANS interface are used as a boundary condition for the RANS solver in the wall region
- ▶ The wall-normal velocity,  $\bar{v}_j^{RANS}$ , is solved for using the pressure at node  $j+1$ .

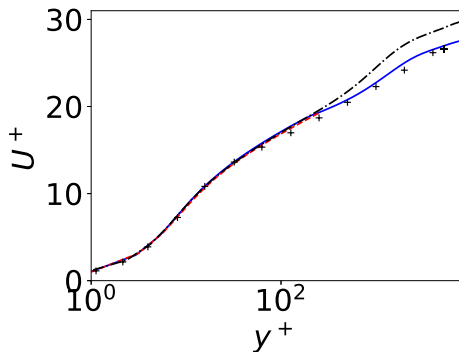
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- ▶ We denote the method NZ S-DES (Non-Zonal approach using Steady RANS coupled to DES)
- ▶ Reynolds number is  $Re_\tau = 8\,000$ .
- ▶ A  $32 \times 96 \times 32$  mesh is used
- ▶  $x_{max} = 3.2$ ,  $z_{max} = 1.6$ , 15% stretching in  $y$  direction

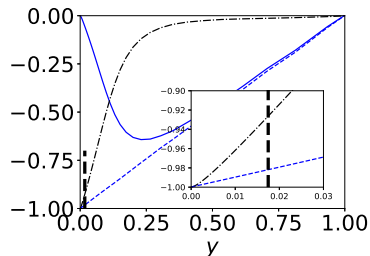
# CHANNEL FLOW: VELOCITY



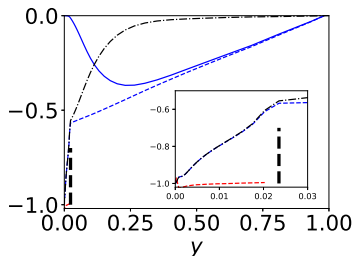
**FIGURE:** Channel flow. NZ S-DES compared with standard *DES*. — : DES solver in NZ S-DES; - - : RANS solver in NZ S-DES; - · - : Standard DES; +: Reichardt's law,  $U^+ = \frac{1}{\kappa} \ln(1 - 0.4y^+) + 7.8 [1 - \exp(-y^+/11) - (y^+/11) \exp(-y^+/3)]$ .



# CHANNEL FLOW: SHEAR STRESSES



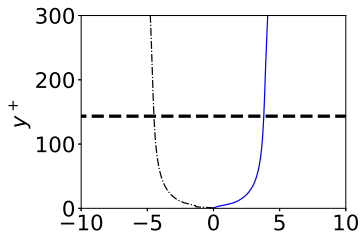
(A) Standard DES.



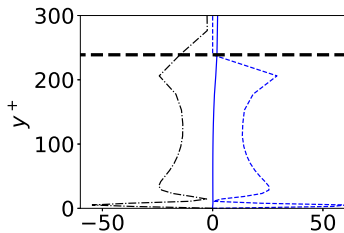
(B) NZ S-DES.

**FIGURE:** Vertical black dashed lines show RANS-LES interface. **—** : resolved; **- · -** : viscous + modeled; **- - -** : total. **- - -** : viscous plus modeled in RANS solver (EARSM) in NZ S-DES.

# CHANNEL FLOW: FORCES



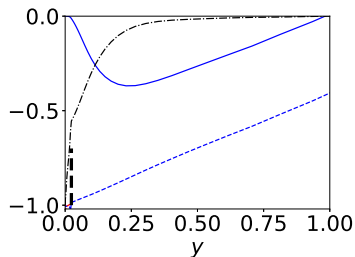
(A) Standard DES.



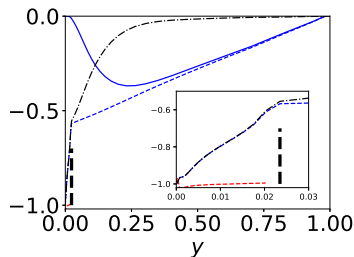
(B) NZ S-DES.

FIGURE: Forces in the  $\langle \bar{u}_1 \rangle$  equation.  $\text{—}$  :  $-\frac{\partial \langle u'v' \rangle}{\partial x_2}$ ;  $\text{---}$  :  $\frac{\partial}{\partial x_2} \left( \langle \nu_{tot} \rangle \frac{\partial \langle \bar{u} \rangle}{\partial x_2} \right)$ ;  $\text{— —}$  : Drift term,  $S_1^{DES}$ .

# SHEAR STRESSES INCLUDING DRIFT TERM



(A) NZ S-DES. **—** : total, including  
 $D(y) = \int_0^y S_1^{DES}(\eta) d\eta$ .



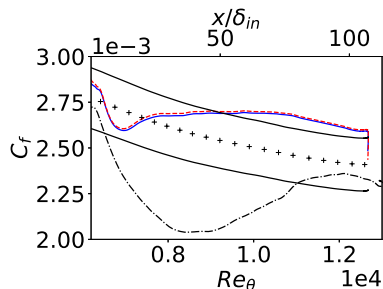
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**FIGURE:** Vertical black dashed lines show RANS-LES interface. **—** : resolved; **- · -** : viscous + modeled; **—** : total. **- - -** : viscous plus modeled in RANS solver (EARSM) in NZ S-DES.

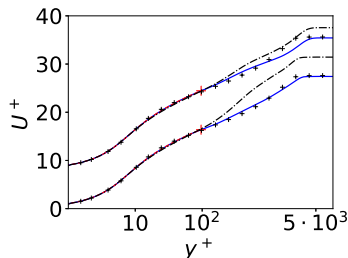
# FLAT-PLATE BOUNDARY LAYER

- ▶  $Re_\theta = 6100$ .
- ▶ Mesh:  $1024 \times 160 \times 64$  cells ( $x, y, z$ )
- ▶  $\Delta z_{in}^+ = 85$  and  $\Delta x_{in}^+ = 280$ .
- ▶ Anisotropic, synthetic turbulence at inlet [3, 1].
- ▶ Commutation terms in the  $k$  and  $\omega$  equations are used at inlet [4, 1].

# SKIN FRICTION AND VELOCITIES



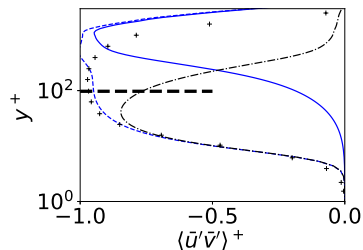
(A) Skinfriction v. momentum thickness. +: DNS; —:  $\pm 6\%$



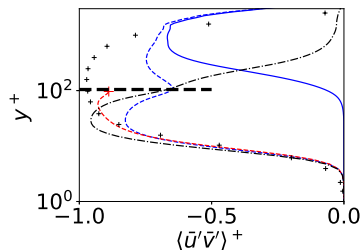
(B) Mean velocity. Lower lines:  $x/\delta_{in} = 8000$ ; upper lines:  $x/\delta_{in} = 10500$ . + DNS at  $Re_\theta = 8000$

FIGURE: — : NZ S-DES; - - - : Standard DES; - - - : RANS solver in NZ S-DES.

# SHEAR STRESSES



(A) Standard DES



(B) NZ S-DES.

**FIGURE:**  $x/\delta_{in} = 75, Re_\theta = 10\,200$ . — : resolved; - - : viscous plus modeled; — — : total; - - : viscous plus modeled in RANS solver (EARSM) in NZ S-DES; + DNS at  $Re_\theta = 8000$

## THIRD TEST CASE: HUMP FLOW

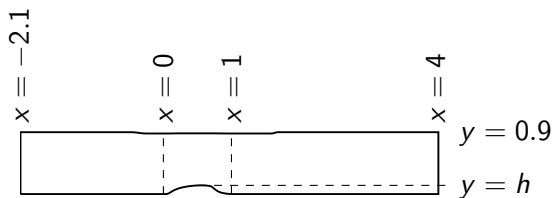
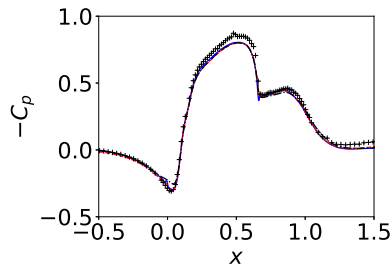


FIGURE: The domain of the hump.  $z_{max} = 0.3$ .

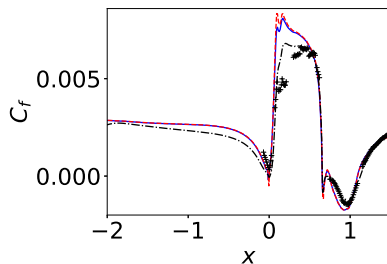
- ▶ The Reynolds number of the hump flow is  $Re_c = 936\,000$ .
- ▶ The mesh has  $648 \times 108 \times 64$  cells ( $x, y, z$ ) with  $\Delta t = 0.0015$
- ▶ Grid from NASA workshop<sup>1</sup>; refined near the inlet and outlet.
- ▶ Inlet is located at  $x = -2.1$  and the outlet at  $x = 4.0$ ,
- ▶ Experiments by [8, 7]

<sup>1</sup>[https://turbmodels.larc.nasa.gov/nasahump\\_val.html](https://turbmodels.larc.nasa.gov/nasahump_val.html)

# PRESSURE AND SKIN FRICTION



(A) Pressure — :  $\pm 6\%$

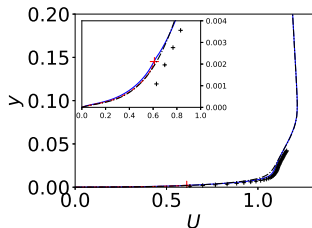


(B) Skin friction.

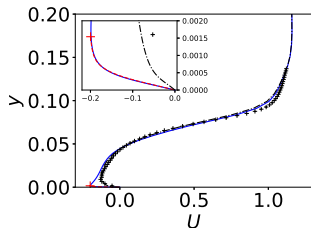
**FIGURE:** Pressure and skin friction. — : NZ S-DES; - - - : Standard DES; - - - : RANS solver in NZ S-DES. +: expts. [8, 7]



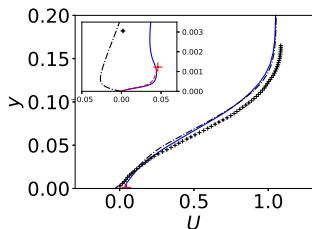
# VELOCITIES



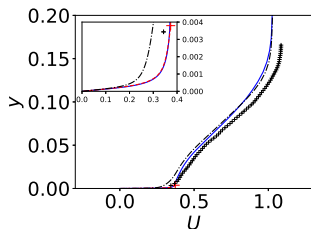
(A)  $x = 0.65$



(B)  $x = 0.80$



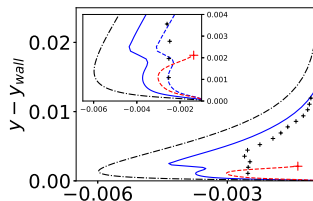
(C)  $x = 1.1$



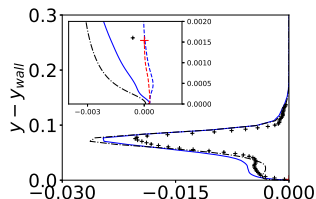
(D)  $x = 1.3$

FIGURE: — : NZ S-DES; - - : Standard DES; - - : RANS solver

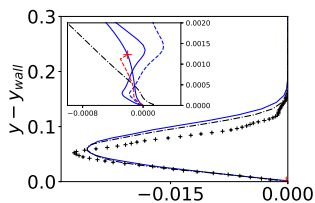
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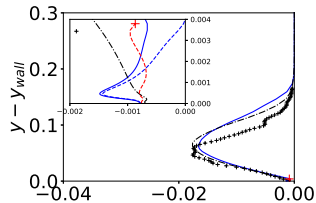
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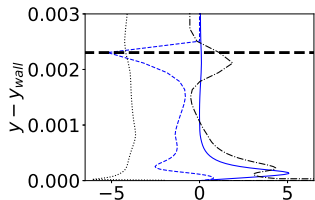
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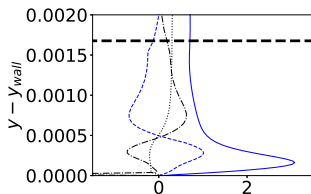
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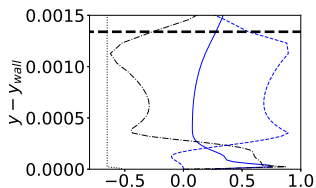
# FORCES



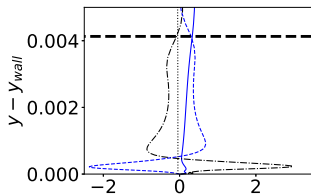
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(D)  $x = 1.3$

FIGURE:  $\text{—}$  :  $-\frac{\partial \langle u'v' \rangle}{\partial y}$ ;  $\text{---}$  :  $\frac{\partial}{\partial y} \left( \langle v_{tot} \rangle \frac{\partial \langle \bar{u} \rangle}{\partial y} \right)$ ;  $\cdots$  :  $-\frac{\partial \langle p \rangle}{\partial x}$ ;  $\text{---}$  :  
 Drift term,  $S_1^{DES}$ .

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- ▶ The 2D RANS solver increase the CPU by **10%**
- ▶ Future development: replace full RANS solver with **boundary-layer solver** prescribing the pressure gradient from the DES solver [2]

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