

Problem 5.1

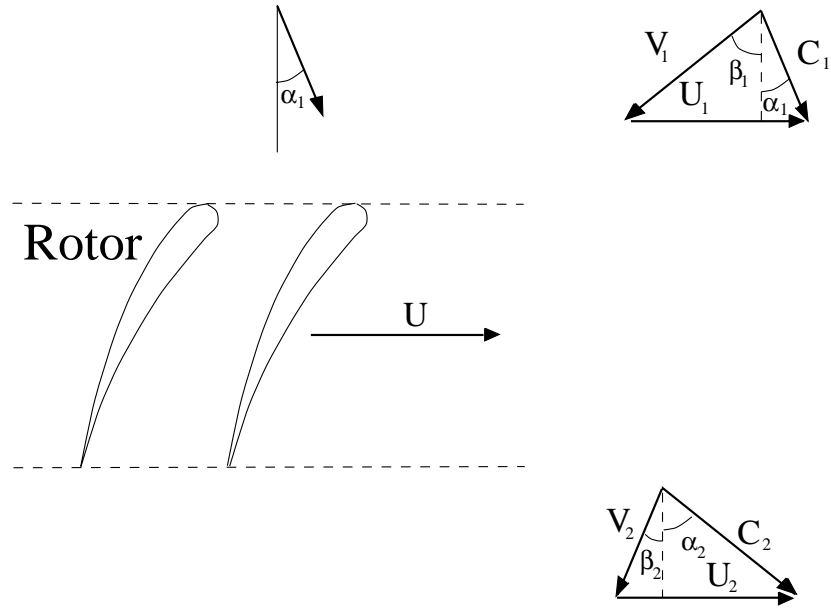


Figure 1: Rotor with air and blade angles

Problem definition: Calculate

- Air angles at root mean and tip
- Degree of reaction at root and tip

Solution: The stage temperature rise is (C.R.S. 194):

$$T_{02} - T_{01} = \frac{\lambda U C_a}{c_p} (\tan \beta_1 - \tan \beta_2) \quad (1)$$

The degree of reaction is (C.R.S. 196):

$$\Lambda = \frac{\text{Static enthalpy rise in rotor}}{\text{Static enthalpy rise in stage}} = \frac{C_a}{2U} (\tan \beta_1 + \tan \beta_2) \quad (2)$$

Evaluated at the mean radius 1 and 2 gives:

$$\tan \beta_1 - \tan \beta_2 = 0.7204$$

$$\tan \beta_1 + \tan \beta_2 = 1.3333$$

\Leftrightarrow

$$\begin{aligned}\beta_1 &= 45.8^\circ \\ \beta_2 &= 17.0^\circ\end{aligned}$$

$\Lambda = 0.50$ designs are symmetrical (C.R.S. 197), i.e.

$$\begin{aligned}\alpha_2 &= \beta_1 = 45.8^\circ \\ \alpha_1 &= \beta_2 = 17.0^\circ\end{aligned}$$

Since this is a free vortex design we must have $C_w r = \text{constant}$. At the mean radius we have:

$$\Delta T_0 = \frac{\lambda U}{c_p} C_a (\tan \alpha_2 - \tan \alpha_1) = \frac{\lambda U}{c_p} (C_{w2} - C_{w1}) = \frac{\lambda U}{c_p} \Delta C_w$$

which gives:

$$\frac{c_p \Delta T_0}{\lambda U} = \frac{1005 \cdot 20}{0.93 \cdot 200} = \Delta C_w = \dots = 108.1 \text{ m/s} \quad (3)$$

where C_{w1} is:

$$C_{w1} = C_a \tan \alpha_1 = 150 \cdot \tan(17.0) = 45.97 \quad (4)$$

3 and 4 give:

$$C_{w2} = \dots = 154.0 \text{ m/s}$$

The free vortex condition for the leading edge gives:

$$\begin{aligned}C_{w1,t} &= \frac{C_{w1,m} r_m}{r_t} = \frac{C_{w1,m} U_m}{U_t} = 36.77 \text{ m/s} \\ C_{w1,r} &= \dots = 61.29 \text{ m/s}\end{aligned}$$

The free vortex condition for the trailing edge gives:

$$\begin{aligned}C_{w2,t} &= \frac{C_{w2,m} r_m}{r_t} = 123.23 \text{ m/s} \\ C_{w2,r} &= \dots = 205.38 \text{ m/s}\end{aligned}$$

At the leading edge we have:

$$\tan \beta_1 = \frac{U - C_{w1}}{C_a} \quad (5)$$

Applied to the root and tip 5 gives:

$$\begin{aligned}\beta_{1,t} &= \dots = 54.87^\circ \\ \beta_{1,r} &= \dots = 30.60^\circ\end{aligned}$$

For the trailing edge we have:

$$\tan\beta_2 = \frac{U - C_w2}{C_a} \quad (6)$$

Applied to the root and tip 6 gives:

$$\begin{aligned}\beta_{2,t} &= \dots = 40.20^\circ \\ \beta_{2,r} &= \dots = -20.26^\circ\end{aligned}$$

The air angles are obtained from (C.R.S. 187):

$$\begin{aligned}\frac{U}{C_a} &= \tan\alpha_1 + \tan\beta_1 \\ \frac{U}{C_a} &= \tan\alpha_2 + \tan\beta_2\end{aligned}$$

which yields (for the leading edge root and tip):

$$\begin{aligned}\alpha_{1,r} &= 22.23^\circ \\ \alpha_{1,t} &= 13.78^\circ\end{aligned}$$

and (for the trailing edge root and tip):

$$\begin{aligned}\alpha_{2,r} &= 53.86^\circ \\ \alpha_{2,t} &= 39.40^\circ\end{aligned}$$

Finally, the degree of reaction is determined by 2:

$$\Lambda_r = \frac{150}{2 \cdot 150}(\tan(30.60) + \tan(-20.26)) = \dots = 0.1111$$

$$\Lambda_t = \frac{150}{2 \cdot 250}(\tan(54.87) + \tan(40.20)) = \dots = 0.6800$$