# Quasi-One-Dimensional Flow

**Governing Equations** 

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## Governing Equations for Quasi-one-dimensional Flow



Figur 1: Quasi-one-dimensional flow - control volume

In the following quasi-one-dimensional flow will be assumed. That means that the cross-section is allowed to vary smoothly but flow quantities varies in one direction only. The equations that are derived will thus describe one-dimensional flow in axisymmetric tubes. Let's assume flow in the x-direction, which means that all flow quantities and the cross-section area will vary with the axial coordinate x.

$$A = A(x), \ \rho = \rho(x), \ u = u(x), \ p = p(x), \ \dots$$

We will further assume steady-state flow, which means that unsteady terms will be zero.

The equations are derived with the starting point in the governing flow equations on integral form

### **Continuity Equation**

Applying the integral form of the continuity equation on the quasi-one-dimensional flow control volume (Fig. 1) gives

$$\underbrace{\frac{d}{dt} \iiint_{\Omega} \rho d\mathscr{V}}_{=0} + \oint_{\partial \Omega} \rho \mathbf{v} \cdot \mathbf{n} dS = 0 \tag{1}$$

$$\rho_1 u_1 A_1 = \rho_2 u_2 A_2 \tag{2}$$

#### Momentum Equation

Applying the integral form of the momentum equation on the quasi-one-dimensional flow control volume (Fig. 1) gives

$$\underbrace{\frac{d}{dt} \iiint_{\Omega} \rho \mathbf{v} d\mathscr{V}}_{=0} + \oiint_{\partial \Omega} \left[ \rho(\mathbf{v} \cdot \mathbf{n}) \mathbf{v} + p \mathbf{n} \right] dS = 0$$
(3)

$$\oint \int_{\partial\Omega} p\mathbf{n} dS = -p_1 A_1 + p_2 A_2 - \int_{A_1}^{A_2} p dA$$

collecting terms

$$\left(\rho_1 u_1^2 + p_1\right) A_1 + \int_{A_1}^{A_2} p dA = \left(\rho_2 u_2^2 + p_2\right) A_2 \tag{4}$$

#### **Energy Equation**

Applying the integral form of the energy equation on the quasi-one-dimensional flow control volume (Fig. 1) gives

$$\underbrace{\frac{d}{dt}\iiint_{\Omega}\rho e_{o}d\mathscr{V}}_{=0} + \oiint_{\partial\Omega}\left[\rho h_{o}(\mathbf{v}\cdot\mathbf{n})\right]dS = 0$$
(5)

$$\oint_{\partial\Omega} \left[\rho h_o(\mathbf{v} \cdot \mathbf{n})\right] dS = -\rho_1 u_1 h_{o_1} A_1 + \rho_2 u_2 h_{o_2} A_2$$

$$\rho_1 u_1 h_{o_1} A_1 = \rho_2 u_2 h_{o_2} A_2$$

Now, using the continuity equation  $\rho_1 u_1 A_1 = \rho_2 u_2 A_2$  gives

$$h_{o_1} = h_{o_2}$$
 (6)

#### **Differential Form**

The integral term appearing the momentum equation is undesired and therefore the governing equations are converted to differential form.

The continuity equation (Eqn. 2) is rewritten in differential form as

$$\rho_1 u_1 A_1 = \rho_2 u_2 A_2 = const$$

$$d(\rho u A) = 0 \tag{7}$$

The momentum equation (Eqn. 4) is rewritten in differential form as

$$\left(\rho_1 u_1^2 + p_1\right) A_1 + \int_{A_1}^{A_2} p dA = \left(\rho_2 u_2^2 + p_2\right) A_2 \Rightarrow d\left[(\rho u^2 + p)A\right] = p dA$$

$$d(\rho u^2 A) + d(pA) = pdA$$

$$ud(\rho uA) + \rho uAdu + Adp + pdA = pdA$$

From the continuity equation we have  $d(\rho uA)$  and thus

$$\rho u \mathcal{A} du + \mathcal{A} dp = 0 \Rightarrow$$

$$dp = -\rho u du \tag{8}$$

which is the momentum equation on differential form. Also referred to as Euler's equation. Finally, the energy equation (Eqn. 2) is rewritten in differential form as

$$h_{o_1} = h_{o_2} = const \Rightarrow dh_o = 0$$

$$h_o = h + \frac{1}{2}u^2 \Rightarrow dh + \frac{1}{2}d(u^2) = 0$$

$$dh + udu = 0 \tag{9}$$

# Summary

Continuity:

$$d(\rho uA) = 0$$

Momentum:

$$dp = -\rho u du$$

Energy:

dh+udu=0

The equations are valid for:

- quasi-one-dimensional flow
- steady state
- all gas models (no gas model assumptions made)
- inviscid flow

It should be noted that equations are exact but they are applied to a physical model that is approximate, i.e., the approximation that flow quantities varies in one dimension with a varying cross-section area. In reality, a variation of cross-section area would imply flow in three dimensions.