

Mechanics of solids and fluids  
-Introduction to continuum mechanics

by

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# Introduction to continuum mechanics

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# 1. Tensors

## 1.1 Index notation

Before introducing concepts of tensor algebra we introduce the index notation. The index notation simplifies writing of quantities as well as equations and will be used in the remaining of this text. There are two types of indices:

- *Free indices* are only used once per quantity and can take the integer values 1, 2 and 3. For example:

$$\begin{aligned}a_i &\Leftrightarrow a_1, a_2 \text{ and } a_3 \\a_i = b_i &\Leftrightarrow a_1 = b_1, a_2 = b_2 \text{ and } a_3 = b_3\end{aligned}$$

Similarly we can have two (or more) free indices

$$\begin{aligned}a_{ij} &\Leftrightarrow a_{11}, a_{12}, a_{13}, a_{21}, a_{22}, a_{23}, a_{31}, a_{32} \text{ and } a_{33} \\a_{ij} = b_{ij} &\Leftrightarrow a_{11} = b_{11}, a_{12} = b_{12}, a_{13} = b_{13}, \dots, a_{32} = b_{32} \text{ and } a_{33} = b_{33}\end{aligned}$$

- *Summation indices* are used twice per term and indicates a summation of that index from 1 to 3. For example:

$$\begin{aligned}a_{ii} &\Leftrightarrow \sum_{i=1}^3 a_{ii} \\a_i b_i &\Leftrightarrow \sum_{i=1}^3 a_i b_i \\a_{ij} b_{ij} &\Leftrightarrow \sum_{i=1}^3 \sum_{j=1}^3 a_{ij} b_{ij}\end{aligned}$$

This sum over repeated indices is often called Einstein's summation convention.

Often these two types of indices are used together. A simple example is the equation system

$$a_i = T_{ij} b_j \Leftrightarrow a_1 = \sum_{j=1}^3 T_{1j} b_j, a_2 = \sum_{j=1}^3 T_{2j} b_j \text{ and } a_3 = \sum_{j=1}^3 T_{3j} b_j$$

Another way to express this equation system is to use *matrices* (in this example two column matrices  $3 \times 1$  and a square matrix  $3 \times 3$ ):

$$\begin{bmatrix} a_1 \\ a_2 \\ a_3 \end{bmatrix} = \begin{bmatrix} T_{11} & T_{12} & T_{13} \\ T_{21} & T_{22} & T_{23} \\ T_{31} & T_{32} & T_{33} \end{bmatrix} \begin{bmatrix} b_1 \\ b_2 \\ b_3 \end{bmatrix} \quad (1.1)$$

which also sometimes is written by using index notation:

$$[a_i] = [T_{ij}] [b_j] \quad (1.2)$$

**Problem 1** Explain the following symbols:  $A_{ii}$ ,  $A_{ijj}$ ,  $A_{ij}$ ,  $a_i A_{ij}$ ,  $c_i b_j A_{ij}$ .  
For each index tell whether it is a summation/dummy index or a free index.

**Problem 2** Use index notation to re-write the following expression:  $f_1 u_1 + f_2 u_2 + f_3 u_3$   
*Answer:*  $f_i u_i$

**Problem 3** Expand  $a_{ijk} b_{ik}$  by giving the terms explicitly.

*Answer:*  $a_{i1k} b_{ik} = \sum_{i=1}^3 \sum_{k=1}^3 a_{i1k} b_{ik} = a_{111} b_{11} + a_{112} b_{12} + a_{113} b_{13} + a_{211} b_{21} + a_{212} b_{22} + a_{213} b_{23} + a_{311} b_{31} + a_{312} b_{32} + a_{313} b_{33},$

$a_{i2k} b_{ik} = \sum_{i=1}^3 \sum_{k=1}^3 a_{i2k} b_{ik} = a_{121} b_{11} + a_{122} b_{12} + a_{123} b_{13} + a_{221} b_{21} + a_{222} b_{22} + a_{223} b_{23} + a_{321} b_{31} + a_{322} b_{32} + a_{323} b_{33},$

$a_{i3k} b_{ik} = \sum_{i=1}^3 \sum_{k=1}^3 a_{i3k} b_{ik} = a_{131} b_{11} + a_{132} b_{12} + a_{133} b_{13} + a_{231} b_{21} + a_{232} b_{22} + a_{233} b_{23} + a_{331} b_{31} + a_{332} b_{32} + a_{333} b_{33}$

**Hand-in assignment 1** (a) Use index notation to re-write the following expression:  
 $b_{11} c_1 d_1 + b_{12} c_2 d_1 + b_{13} c_3 d_1 + b_{21} c_1 d_2 + b_{22} c_2 d_2 + b_{23} c_3 d_2 + b_{31} c_1 d_3 + b_{32} c_2 d_3 + b_{33} c_3 d_3.$   
(b) Expand  $a_{ij} = b_{mj} c_{in} d_{mn}$  by giving the terms explicitly.

**Matlab example 1** An example of using Matlab commands for matrix definitions (for  $T$  and  $b$ ) and multiplication  $a_i = T_{ij} b_j$  is given below:

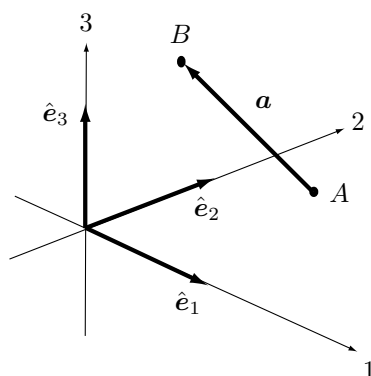
```
>> T=[1 2 3; 4 5 6; 7 8 9];
>> b=[1 2 3]';
>> a=T*b
```

a =

14

32

50

Figure 1.1: Illustration of vector  $\mathbf{a}$ .

## 1.2 Vectors

For further reading see Reddy 2.2 *Vector Algebra*.

### Orthonormal base vectors

To describe many physical quantities (such as force, displacement, velocity) both magnitude and direction must be given. Hence, these quantities can be described by vectors (1st order tensors) in a 3-dimensional Euclidean space. By introducing a set of right-handed orthonormal basis vectors  $\{\hat{\mathbf{e}}_1, \hat{\mathbf{e}}_2, \hat{\mathbf{e}}_3\}$  any vector  $\mathbf{a} = \overrightarrow{AB}$  can be expressed as a linear combination these basis vectors,  $\hat{\mathbf{e}}_i$ :

$$\mathbf{a} = a_1 \hat{\mathbf{e}}_1 + a_2 \hat{\mathbf{e}}_2 + a_3 \hat{\mathbf{e}}_3 = a_j \hat{\mathbf{e}}_j. \quad (1.3)$$

as shown in figure 1.1. The coefficients  $a_i$  or  $(a_1, a_2, a_3)$  are the *components* of  $\mathbf{a}$  with respect to the basis  $\hat{\mathbf{e}}_i$ . The length (=Euclidean norm) of a vector  $\mathbf{a}$  is denoted  $a$  or  $|\mathbf{a}|$ . For normalized vectors (describing only direction) the following notation is introduced:

$$\hat{\mathbf{e}}_a = \frac{\mathbf{a}}{a}, \quad (1.4)$$

whereby a vector  $\mathbf{a}$  can be written as  $\mathbf{a} = a \hat{\mathbf{e}}_a$ . Examples of normalized vectors are the basis vectors  $\{\hat{\mathbf{e}}_1, \hat{\mathbf{e}}_2, \hat{\mathbf{e}}_3\}$ .

### Scalar product

To each pair of vectors  $\mathbf{a}$  and  $\mathbf{b}$  there corresponds a real number  $\mathbf{a} \cdot \mathbf{b}$ , called the scalar product. The scalar product can be obtained as (see figure 1.2):

$$\mathbf{a} \cdot \mathbf{b} = ab \cos \theta \quad (1.5)$$

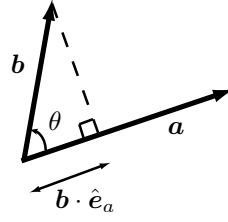


Figure 1.2: Illustration of scalar product.

whereby it is possible to express the length of a vector  $a = |\mathbf{a}|$  as follows

$$a = |\mathbf{a}| = \sqrt{\mathbf{a} \cdot \mathbf{a}} \quad (1.6)$$

By now applying the scalar product between the orthonormal basis vectors  $\{\hat{\mathbf{e}}_1, \hat{\mathbf{e}}_2, \hat{\mathbf{e}}_3\}$ , the following results are obtained

$$\hat{\mathbf{e}}_i \cdot \hat{\mathbf{e}}_j = \delta_{ij} \quad (1.7)$$

where

$$\delta_{ij} = \begin{cases} 1 & \text{when } i = j \\ 0 & \text{when } i \neq j \end{cases} \quad (1.8)$$

The symbol  $\delta_{ij}$  is called the *Kronecker delta symbol*. The scalar product between vectors is a bilinear operator and has the following properties:

$$\begin{cases} \mathbf{a} \cdot (\alpha \mathbf{b} + \beta \mathbf{c}) &= \alpha \mathbf{a} \cdot \mathbf{b} + \beta \mathbf{a} \cdot \mathbf{c} \\ (\alpha \mathbf{a} + \beta \mathbf{b}) \cdot \mathbf{c} &= \alpha \mathbf{a} \cdot \mathbf{c} + \beta \mathbf{b} \cdot \mathbf{c} \end{cases}$$

where  $\alpha$  and  $\beta$  are scalars. These properties can now be used to show that the scalar product between two vectors  $\mathbf{a}$  and  $\mathbf{b}$  may be written as

$$\mathbf{a} \cdot \mathbf{b} = a_i b_j \hat{\mathbf{e}}_i \cdot \hat{\mathbf{e}}_j = a_i b_j \delta_{ij} = a_i b_i \quad (1.9)$$

and that the scalar product is commutative i.e.  $\mathbf{a} \cdot \mathbf{b} = \mathbf{b} \cdot \mathbf{a}$ . An observation is that the components  $a_i$  of a vector  $\mathbf{a}$  can be extracted by scalar multiplication with corresponding base vectors  $\hat{\mathbf{e}}_i$ , i.e.

$$a_i = \hat{\mathbf{e}}_i \cdot \mathbf{a} \quad (1.10)$$

## Vector product

Another product that is useful is the vector product  $\mathbf{a} \times \mathbf{b}$ , that is illustrated in Figure 1.3. The result is a vector that is orthogonal to the plane spanned by  $\mathbf{a}$  and  $\mathbf{b}$  (with a right-handed system) and has a length

$$|\mathbf{a} \times \mathbf{b}| = ab \sin \theta. \quad (1.11)$$



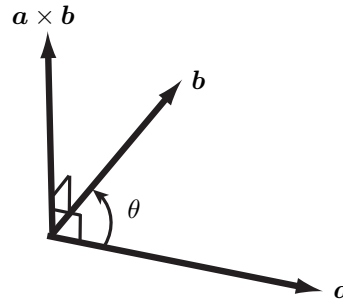


Figure 1.3: Illustration of vector product.

By applying the vector product to the orthonormal basis vectors  $\{\hat{e}_1, \hat{e}_2, \hat{e}_3\}$ , the following results are obtained

$$\hat{e}_i \times \hat{e}_j = e_{ijk} \hat{e}_k, \quad (1.12)$$

where

$$e_{ijk} = \begin{cases} 1 & \text{when } ijk = 123, 231 \text{ or } 312 \\ -1 & \text{when } ijk = 321, 213 \text{ or } 132 \\ 0 & \text{otherwise} \end{cases} \quad (1.13)$$

The symbol  $e_{ijk}$  is called the *permutation symbol*. The vector product is bilinear i.e.

$$\begin{cases} \mathbf{a} \times (\alpha \mathbf{b} + \beta \mathbf{c}) = \alpha \mathbf{a} \times \mathbf{b} + \beta \mathbf{a} \times \mathbf{c} \\ (\alpha \mathbf{a} + \beta \mathbf{b}) \times \mathbf{c} = \alpha \mathbf{a} \times \mathbf{c} + \beta \mathbf{b} \times \mathbf{c} \end{cases}$$

whereby the vector product between two arbitrary vectors becomes

$$\mathbf{a} \times \mathbf{b} = a_i b_j \hat{e}_i \times \hat{e}_j = a_i b_j e_{ijk} \hat{e}_k. \quad (1.14)$$

An observation is the relation  $\mathbf{a} \times \mathbf{b} = -\mathbf{b} \times \mathbf{a}$  which is found by using the properties of the permutation symbol. The permutation symbol and the Kronecker's delta symbol are linked by the so-called  $\epsilon$ - $\delta$  identity:

$$e_{ijm} e_{klm} = \delta_{ik} \delta_{jl} - \delta_{il} \delta_{jk}. \quad (1.15)$$

### Open product

Open product (also called outer product) between two vectors  $\mathbf{a}$  and  $\mathbf{b}$  results in a 2nd order tensor  $\mathbf{T}$  (also called dyad) as follows

$$\mathbf{a} \mathbf{b} = a_i \hat{e}_i b_j \hat{e}_j = a_i b_j \hat{e}_i \hat{e}_j = T_{ij} \hat{e}_i \hat{e}_j = \mathbf{T} \quad (1.16)$$

The open product is bilinear but not commutative i.e.  $\mathbf{a} \mathbf{b} \neq \mathbf{b} \mathbf{a}$  in general. "2nd order tensors will be exploited further in Section 1.3.

### Column matrix representation of components

In a given coordinate system defined by the basis vectors  $\{\hat{e}_1, \hat{e}_2, \hat{e}_3\}$  a vector  $\mathbf{a}$  or rather the vector components  $a_i$  can be collected in a column matrix as follows

$$[\mathbf{a}] = [a_i] = [a_1 \ a_2 \ a_3]^T \quad (1.17)$$

An example is the base vector  $\hat{e}_1$  that is represented by the following column matrix

$$[\hat{e}_1] = [1 \ 0 \ 0]^T \quad (1.18)$$

Therefore, the scalar multiplication between two vectors can be obtained as

$$\mathbf{a} \cdot \mathbf{b} = a_i b_i = [\mathbf{a}]^T [\mathbf{b}]. \quad (1.19)$$

### Coordinate system transformation

A vector must be invariant with respect to coordinate system. Assume two different sets of orthonormal basis vectors  $\{\hat{e}_1, \hat{e}_2, \hat{e}_3\}$  and  $\{\hat{e}'_1, \hat{e}'_2, \hat{e}'_3\}$ . The vector  $\mathbf{b}$  can then be written as

$$\mathbf{b} = b_i \hat{e}_i = b'_i \hat{e}'_i \quad (1.20)$$

The components  $b'_i$  can be extracted from  $\mathbf{b}$  as

$$b'_i = \hat{e}'_i \cdot \mathbf{b} = \hat{e}'_i \cdot b_j \hat{e}_j = \hat{e}'_i \cdot \hat{e}_j b_j \quad (1.21)$$

In matrix notation this can be written

$$[b'_i] = [l_{ij}] [b_j] = [\hat{e}'_i \cdot \hat{e}_j] [b_j] \quad (1.22)$$

where the transformation matrix  $[l_{ij}]$  is orthogonal, i.e.  $[l_{ij}]^T = [l_{ij}]^{-1}$ .

**Problem 4** Determine the unit length vector along  $\mathbf{a} = 4\hat{e}_1 + 6\hat{e}_2 - 12\hat{e}_3$ .

*Answer:*  $\hat{\mathbf{a}} = 2/7\hat{e}_1 + 3/7\hat{e}_2 - 6/7\hat{e}_3$

**Problem 5** Compute the projection of the vector  $\mathbf{a} = 4\hat{e}_1 + 6\hat{e}_2 - 12\hat{e}_3$  on the line defined by the vector  $\mathbf{b} = 1\hat{e}_1 + 1\hat{e}_2 + 1\hat{e}_3$

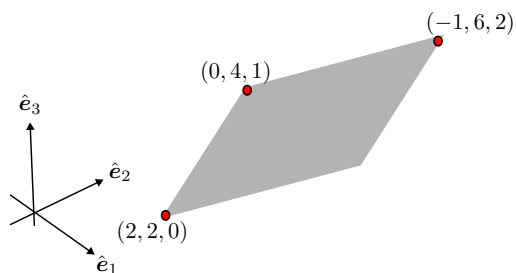
*Answer:*  $-2/\sqrt{3}$

**Problem 6** Expand the following expressions of the Kronecker delta  $\delta_{ij}$ :

$\delta_{ij}\delta_{ij}, \delta_{ij}\delta_{jk}\delta_{ki}, \delta_{ij}A_{ik}$

*Answers:* 3, 3,  $A_{jk}$ .

**Problem 7** Compute the unit normal to the plane

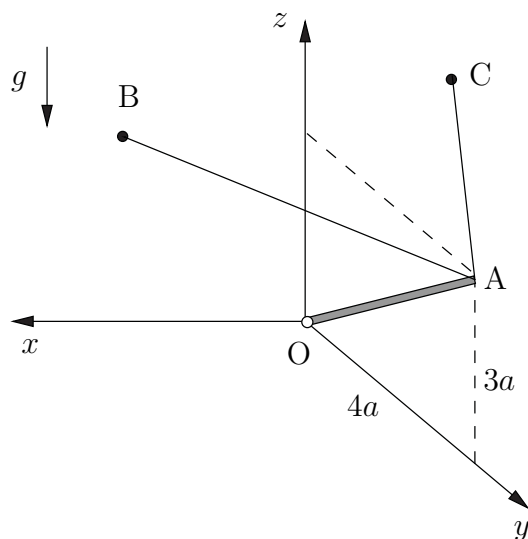


*Answer:*  $\hat{\mathbf{n}} \approx -0.45 \hat{\mathbf{e}}_2 + 0.89 \hat{\mathbf{e}}_3$

**Problem 8** Prove that for three arbitrary vectors  $\mathbf{a}$ ,  $\mathbf{b}$  and  $\mathbf{c}$  the following relation holds:

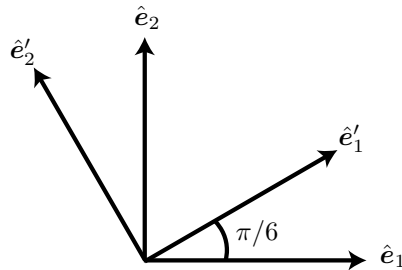
$$\mathbf{a} \times (\mathbf{b} \times \mathbf{c}) = (\mathbf{a} \cdot \mathbf{c}) \mathbf{b} - (\mathbf{a} \cdot \mathbf{b}) \mathbf{c}$$

**Hand-in assignment 2** A thin rigid bar OA with mass  $m$  and length  $5a$  is attached without friction in a joint at O. The bar is kept in equilibrium by two light cables AB and AC acc to the figure. The cable AB is attached to the bar at B with coordinates  $(3a; 0; 3a)$  and the cable AC is attached to the bar at C with coordinates  $(-a; 2a; 5a)$ .



Determine the forces in the cables AB and AC at equilibrium ( $g$  is the acceleration of gravity in the negative  $z$  direction).

**Hand-in assignment 3** Give the component of the vector  $\mathbf{a}$  in the rotated coordinate system  $\{\hat{e}'_i\}$ . This coordinate system is obtained from the coordinate system  $\{\hat{e}_i\}$  by rotating around the  $\hat{e}_3$  axis according to the figure.



The components of the vector  $\mathbf{a}$  in the coordinate system  $\{\hat{e}_i\}$  are given as  $[-1 \ 4 \ 3]^T$ .

**Matlab example 2** Example of Matlab input file to define  $e_{ijk}$ -operator and vector product  $c_k = a_i b_j e_{ijk}$ :

```
%definition of permutation symbol
perm=zeros(3,3,3);
for i=1:3
    for j=1:3
        for k=1:3
            %%
            if ( (i==1) & (j==2) & (k==3) ) | ((i==2) & (j==3) & (k==1)) | ...
                ((i==3) & (j==1) & (k==2))
                perm(i,j,k)=1;
            elseif ( (i==3) & (j==2) & (k==1) ) | ((i==2) & (j==1) & (k==3)) | ...
                ((i==1) & (j==3) & (k==2))
                perm(i,j,k)=-1;
            end
            %%
        end
    end
end
%computation of vector product c_k= a_i b_j perm_ijk
a=[1 2 3]';
b=[4 5 6]';
c=zeros(3,1);
for k=1:3
```

```

c(k)=0;
for i=1:3
    for j=1:3
        c(k)=c(k,1)+a(i)*b(j)*perm(i,j,k);
    end
end
end
end

```

### 1.3 2nd order tensors

For further reading see Reddy 2.5 *Tensors*.

#### Representation of 2nd order tensors

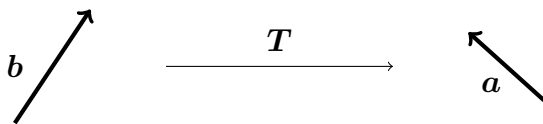
2nd order tensors are physical quantities that describe how vectors change with e.g. direction and position in space. Examples of 2nd order tensors that we will explore later are the stress tensor, strain tensor, velocity gradient and the deformation gradient. A 2nd order tensor  $\mathbf{T}$  is represented in a orthonormal coordinate system  $\{\hat{\mathbf{e}}_1, \hat{\mathbf{e}}_2, \hat{\mathbf{e}}_3\}$  as

$$\mathbf{T} = T_{ij} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_j, \quad (1.23)$$

where  $T_{ij}$  are the nine *components* of  $\mathbf{T}$  and  $\hat{\mathbf{e}}_i \hat{\mathbf{e}}_j$  are the *base dyads*. The base dyads  $\hat{\mathbf{e}}_i \hat{\mathbf{e}}_j$  are 2nd order tensors themselves and  $\mathbf{T}$  is built up by a linear combination of these scaled by the components  $T_{ij}$ .

#### Scalar product between 2nd order tensors and vectors

Now consider a linear transformation of a vector  $\mathbf{b}$  into  $\mathbf{a}$



This may be written symbolically by introducing a scalar product as

$$\mathbf{a} = \mathbf{T} \cdot \mathbf{b} \quad (1.24)$$

where the linear operator  $\mathbf{T}$  is a second-order tensor. Before proceeding we define the scalar product between a base dyad and a base vector as:

$$(\hat{\mathbf{e}}_i \hat{\mathbf{e}}_j) \cdot \hat{\mathbf{e}}_k = \hat{\mathbf{e}}_i (\hat{\mathbf{e}}_j \cdot \hat{\mathbf{e}}_k) = \delta_{jk} \hat{\mathbf{e}}_i \quad \text{and} \quad \hat{\mathbf{e}}_i \cdot (\hat{\mathbf{e}}_j \hat{\mathbf{e}}_k) = (\hat{\mathbf{e}}_i \cdot \hat{\mathbf{e}}_j) \hat{\mathbf{e}}_k = \delta_{ij} \hat{\mathbf{e}}_k \quad (1.25)$$

The scalar product between a 2nd order tensor and a vector is assumed to be bilinear. If we use index notations then such a scalar product can be written as:

$$\mathbf{a} = \mathbf{T} \cdot \mathbf{b} = T_{ij} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_j \cdot b_k \hat{\mathbf{e}}_k = T_{ij} b_k \hat{\mathbf{e}}_i \delta_{jk} = \underbrace{T_{ij} b_j}_{a_i} \hat{\mathbf{e}}_i = a_i \hat{\mathbf{e}}_i \quad (1.26)$$

Often we omit the basis and simply write the relation between the components, i.e.

$$a_i = T_{ij} b_j \quad (1.27)$$

It is then implicitly assumed that the same basis vectors are used for all variables. If we switch the order of the vector and the 2nd order tensor in the scalar product

$$\mathbf{c} = \mathbf{b} \cdot \mathbf{T} = b_i \hat{\mathbf{e}}_i \cdot T_{jk} \hat{\mathbf{e}}_j \hat{\mathbf{e}}_k = b_i T_{jk} \delta_{ij} \hat{\mathbf{e}}_k = \underbrace{b_j T_{jk}}_{c_k} \hat{\mathbf{e}}_k = c_k \hat{\mathbf{e}}_k \quad (1.28)$$

or in short  $c_k = b_j T_{jk}$ . A consequence of these results is that:

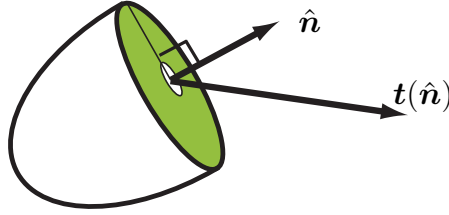
$$\mathbf{T} \cdot \mathbf{b} = \mathbf{b} \cdot \mathbf{T}^T$$

where the transpose of the tensor is defined as  $\mathbf{T}^T = T_{ji} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_j$ .

By using this scalar multiplication twice it is possible to find the components of a 2nd order tensor  $T_{ij}$  by

$$\hat{\mathbf{e}}_i \cdot \mathbf{T} \cdot \hat{\mathbf{e}}_j = T_{ij} \quad (1.29)$$

An example of a 2nd order tensor is the stress tensor  $\boldsymbol{\sigma}$  from which the traction vector  $\mathbf{t}(\hat{\mathbf{n}})$  can be obtained as  $\mathbf{t}(\hat{\mathbf{n}}) = \hat{\mathbf{n}} \cdot \boldsymbol{\sigma}$ .



A special 2nd order tensor is the identity tensor  $\boldsymbol{\delta}$ :

$$\boldsymbol{\delta} = \delta_{ij} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_j = \hat{\mathbf{e}}_i \hat{\mathbf{e}}_i \quad (1.30)$$

with the property that it does not transform a vector  $\mathbf{a}$  when scalar multiplied with  $\boldsymbol{\delta}$ , i.e.  $\boldsymbol{\delta} \cdot \mathbf{a} = \mathbf{a} \cdot \boldsymbol{\delta} = \mathbf{a}$  or written by using only the components  $\delta_{ij} a_j = a_j \delta_{ji} = a_i$ .

### Multiplication between 2nd order tensors

Scalar multiplication (also called single contraction) between two base dyads is defined as

$$(\hat{\mathbf{e}}_i \hat{\mathbf{e}}_j) \cdot (\hat{\mathbf{e}}_k \hat{\mathbf{e}}_l) = \hat{\mathbf{e}}_i (\hat{\mathbf{e}}_j \cdot \hat{\mathbf{e}}_k) \hat{\mathbf{e}}_l = \hat{\mathbf{e}}_i \delta_{jk} \hat{\mathbf{e}}_l = \delta_{jk} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_l \quad (1.31)$$

This scalar multiplication is assumed to be bilinear and therefore the scalar multiplication of two 2nd order  $\mathbf{T}$  and  $\mathbf{U}$  can be written as

$$\mathbf{T} \cdot \mathbf{U} = T_{ij} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_j \cdot U_{kl} \hat{\mathbf{e}}_k \hat{\mathbf{e}}_l = T_{ij} U_{kl} \hat{\mathbf{e}}_i \delta_{jk} \hat{\mathbf{e}}_l = \underbrace{T_{ij} U_{jl}}_{V_{il}} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_l = \mathbf{V} \quad (1.32)$$

or in terms of components  $T_{ij} U_{jk} = V_{ik}$ . Further, by applying the transpose operator to such a product it is straightforward to show that

$$(\mathbf{A} \cdot \mathbf{B})^T = \mathbf{B}^T \cdot \mathbf{A}^T. \quad (1.33)$$

Similarly, if we introduce the double contraction operator between two base dyads as

$$(\hat{\mathbf{e}}_i \hat{\mathbf{e}}_j) : (\hat{\mathbf{e}}_k \hat{\mathbf{e}}_l) = (\hat{\mathbf{e}}_i \cdot \hat{\mathbf{e}}_k) (\hat{\mathbf{e}}_j \cdot \hat{\mathbf{e}}_l) = \delta_{ik} \delta_{jl} \quad (1.34)$$

and assume bilinearity of that operator. Then double contraction between two 2nd order  $\mathbf{T}$  and  $\mathbf{U}$  results in a scalar  $\alpha$  and is obtained as

$$\mathbf{T} : \mathbf{U} = T_{ij} U_{ij} = \alpha \quad (1.35)$$

### Symmetric and skew-symmetric 2nd order tensors

As introduced earlier the transpose  $\mathbf{T}^T$  of a 2nd order tensor  $\mathbf{T}$  is defined as follows:

$$\mathbf{T}^T = T_{ij} \hat{\mathbf{e}}_j \hat{\mathbf{e}}_i = T_{ji} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_j \quad (1.36)$$

Many second-order tensors in mechanics are symmetric which means that the tensor and its transpose are equal e.g.  $\mathbf{T}^T = \mathbf{T}$  or in components  $T_{ij} = T_{ji}$ . Another type of tensors is the skew-symmetric second-order tensors. These have the property that the transpose of the tensor is equal to the tensor with a minus sign, e.g.,  $\mathbf{T}^T = -\mathbf{T}$  or  $T_{ij} = -T_{ji}$ . Clearly, for such a tensor the diagonal elements (in a matrix representation) must be equal to zero.

### Matrix representation of components

Equation (1.27) represents a linear system of equations between vector and 2nd order tensor components. This system of equations can be written in matrix notation as

$$[a_1 \ a_2 \ a_3]^T = \begin{bmatrix} T_{11} & T_{12} & T_{13} \\ T_{21} & T_{22} & T_{23} \\ T_{31} & T_{32} & T_{33} \end{bmatrix} [b_1 \ b_2 \ b_3]^T.$$

This shows why second-order tensors often are represented by  $3 \times 3$  matrices. Note that it is only the components of  $\mathbf{a}$ ,  $\mathbf{b}$  and  $\mathbf{T}$  in the coordinate system  $\{\hat{\mathbf{e}}_1, \hat{\mathbf{e}}_2, \hat{\mathbf{e}}_3\}$  that are represented by the matrices. If the coordinate system would be changed then also these components would change.

### Inverse of a 2nd order tensor

If we assume that the tensor  $\mathbf{T}$  gives the linear transformation  $\mathbf{a} = \mathbf{T} \cdot \mathbf{b}$  between the two vectors  $\mathbf{b}$  and  $\mathbf{a}$ . Then we can introduce the inverse  $\mathbf{T}^{-1}$  of this transformation as  $\mathbf{b} = \mathbf{T}^{-1} \cdot \mathbf{a}$ . If we express these two relations in components

$$a_i = T_{ij} b_j \quad \text{and} \quad b_i = T_{ij}^{-1} a_j \quad (1.37)$$

then it is obvious that the components of the  $\mathbf{T}^{-1}$  can be found using standard matrix inversion i.e.

$$[T_{ij}^{-1}] = [T_{ij}]^{-1} \quad (1.38)$$

Hence, standard rules for matrix inversion apply also for tensor components such as  $(\mathbf{T} \cdot \mathbf{U})^{-1} = \mathbf{U}^{-1} \cdot \mathbf{T}^{-1}$ .

### Coordinate system transformation

A 2nd order tensor is invariant with respect to coordinate system. Assume two different sets of orthonormal basis vectors  $\{\hat{\mathbf{e}}_1, \hat{\mathbf{e}}_2, \hat{\mathbf{e}}_3\}$  and  $\{\hat{\mathbf{e}}'_1, \hat{\mathbf{e}}'_2, \hat{\mathbf{e}}'_3\}$ . The 2nd order tensor  $\mathbf{T}$  can then be written with either basis vectors:

$$\mathbf{T} = T_{ij} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_j = T'_{ij} \hat{\mathbf{e}}'_i \hat{\mathbf{e}}'_j \quad (1.39)$$

The components  $T'_{ij}$  can be extracted from  $\mathbf{T}$  as

$$T'_{ij} = \hat{\mathbf{e}}'_i \cdot \mathbf{T} \cdot \hat{\mathbf{e}}'_j = \hat{\mathbf{e}}'_i \cdot \hat{\mathbf{e}}_k T_{kl} \hat{\mathbf{e}}_l \cdot \hat{\mathbf{e}}'_j \quad (1.40)$$

In matrix notation this can be written

$$[T'_{ij}] = [l_{ik}] [T_{kl}] [l_{jl}]^T \quad (1.41)$$

where the orthogonal transformation matrix  $[l_{ij}]$  was defined in (1.22).

### Higher order tensors

It is possible to construct tensors of any order (or rank) as follows:

$$\mathbf{A} = A_{ijk\dots} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_j \hat{\mathbf{e}}_k \cdots$$

In particular, fourth-order tensors are frequently used to, for example, give the relation (material behavior) between the second-order tensors stress and strain.



**Problem 9** If  $\mathbf{a}$  and  $\mathbf{b}$  are vectors and  $\mathbf{A}$  and  $\mathbf{B}$  are 2nd order tensors show that

a)  $(\mathbf{a} \cdot \mathbf{A}) \cdot \mathbf{b} = \mathbf{a} \cdot (\mathbf{A} \cdot \mathbf{b})$

b)  $(\mathbf{A} \cdot \mathbf{B})^T = \mathbf{B}^T \cdot \mathbf{A}^T$

c)  $(\mathbf{A} \cdot \mathbf{a}) \cdot (\mathbf{B} \cdot \mathbf{b}) = \mathbf{a} \cdot (\mathbf{A}^T \cdot \mathbf{B}) \cdot \mathbf{b}$

**Problem 10** The components of the 2nd order tensors and vector are given as:

$$[A_{ij}] = \begin{bmatrix} 1 & 2 & 0 \\ 2 & 3 & 4 \\ 0 & 4 & 2 \end{bmatrix}, \quad [B_{ij}] = \begin{bmatrix} 3 & 0 & 0 \\ 0 & 3 & 1 \\ 0 & 1 & 2 \end{bmatrix}, \quad [a_i] = \begin{bmatrix} 2 \\ 3 \\ 1 \end{bmatrix}, \quad [b_i] = \begin{bmatrix} 1 \\ -1 \\ 2 \end{bmatrix}$$

Compute

a)  $\mathbf{A} \cdot \mathbf{a}$

b)  $\mathbf{a} \cdot \mathbf{b}$

c)  $\mathbf{A} : \mathbf{B}$

d)  $\mathbf{A} : (\mathbf{a} \mathbf{b})$

*Answers:* a)  $[8, 7, 4]^T$ , b) 1, c) 24, d) 19.

**Problem 11** Show that  $A_{ij} = e_{ijk}a_k$  is skew-symmetric (i.e.  $A_{ji} = -A_{ij}$ ).

**Problem 12** If  $A_{ij}$  is symmetric and  $B_{ij}$  is skew-symmetric. Show that  $A_{ij}B_{ij} = 0$ .

**Problem 13** 2.10 a in Reddy

**2.10** Determine the transformation matrix relating the orthonormal basis vectors  $(\hat{\mathbf{e}}_1, \hat{\mathbf{e}}_2, \hat{\mathbf{e}}_3)$  and  $(\hat{\mathbf{e}}'_1, \hat{\mathbf{e}}'_2, \hat{\mathbf{e}}'_3)$ , when  $\hat{\mathbf{e}}'_i$  are given by

(a)  $\hat{\mathbf{e}}'_1$  is along the vector  $\hat{\mathbf{e}}_1 - \hat{\mathbf{e}}_2 + \hat{\mathbf{e}}_3$  and  $\hat{\mathbf{e}}'_2$  is perpendicular to the plane  $2x_1 + 3x_2 + x_3 - 5 = 0$ .

*Answer:*

$$\begin{pmatrix} \hat{\mathbf{e}}'_1 \\ \hat{\mathbf{e}}'_2 \\ \hat{\mathbf{e}}'_3 \end{pmatrix} = \frac{1}{\sqrt{42}} \begin{pmatrix} \sqrt{14} & -\sqrt{14} & \sqrt{14} \\ 2\sqrt{3} & 3\sqrt{3} & \sqrt{3} \\ -4 & 1 & 5 \end{pmatrix} \begin{pmatrix} \hat{\mathbf{e}}_1 \\ \hat{\mathbf{e}}_2 \\ \hat{\mathbf{e}}_3 \end{pmatrix}$$

**Hand-in assignment 4** Given the Sherman-Morrison's formula:

$$A_{ij} = \delta_{ij} + \alpha u_i v_j \quad \text{then} \quad A_{ij}^{-1} = \delta_{ij} - \frac{\alpha}{1 + \alpha u_k v_k} u_i v_j$$

Show that by using Sherman-Morrison's formula if  $A_{ij} = B_{ij} + \alpha u_i v_j$  then

$$A_{ij}^{-1} = B_{ij}^{-1} - \frac{\alpha}{1 + \alpha v_k B_{kl}^{-1} u_l} B_{im}^{-1} u_m v_n B_{nj}^{-1}$$

**Matlab example 3** An example of using Matlab commands for matrix definitions (for  $A$  and  $B$ ) and computing the contraction  $c = A_{ij} B_{ij}$  is given below:

```
A=[1 2 3; 4 5 6; 7 8 9]; B=[3 2 1; 6 5 4; 9 8 7];
```

```
>> c=sum(sum(A.*B))
```

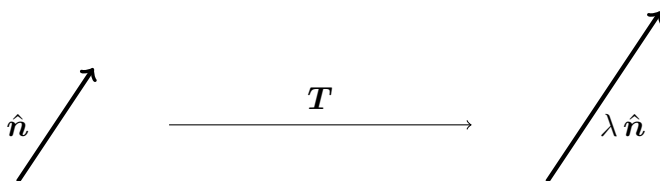
```
c =
```

```
273
```

## 1.4 Principal values and principal directions

For further reading see Reddy 2.5.5 *Eigenvalues and Eigenvectors of Tensors*.

A second-order tensor can be, as discussed above, thought of a linear transformation between vectors, i.e.  $\mathbf{a} = \mathbf{T} \cdot \mathbf{b}$ . Important properties of a second-order tensor are its eigenvectors (principal directions) and eigenvalues (principal values). Eigenvectors are defined as vectors that do not rotate upon transformation with the second-order tensor. If  $\hat{\mathbf{n}}$  now is an eigenvector to  $\mathbf{T}$  this can be illustrated as



This can be written as

$$\lambda \hat{\mathbf{n}} = \mathbf{T} \cdot \hat{\mathbf{n}} \quad \text{or} \quad \lambda \hat{n}_i = T_{ij} \hat{n}_j. \quad (1.42)$$

The eigenvectors  $\hat{\mathbf{n}}$  are chosen to be of unit length whereby it is possible to identify the length of the vector  $\mathbf{T} \cdot \hat{\mathbf{n}}$  as the corresponding eigenvalues  $\lambda$ . The way to find the eigenvalues and eigenvectors is to rewrite (1.42) as

$$(\lambda \boldsymbol{\delta} - \mathbf{T}) \cdot \hat{\mathbf{n}} = \mathbf{0} \quad \text{or} \quad (\lambda \delta_{ij} - T_{ij}) \hat{n}_j = 0_i. \quad (1.43)$$

A trivial solution to this equation is that  $\hat{\mathbf{n}} = \mathbf{0}$ . However, it is possible to find non-trivial solution if  $(\lambda \boldsymbol{\delta} - \mathbf{T})$  is non-invertible. From linear algebra we know that then the determinant of the matrix  $[\lambda \boldsymbol{\delta} - \mathbf{T}]$  must be zero, i.e.

$$\det(\mathbf{T} - \lambda \boldsymbol{\delta}) = 0 \quad (1.44)$$

which is called the characteristic equation. An important theorem from linear algebra is the spectral theorem which states that for symmetric matrices the eigenvalues are real and the eigenvectors are orthogonal. In the current course we will only consider eigenvalues and eigenvectors for symmetric second order tensors (i.e. stress, strain, etc) and for such a tensor the characteristic equation can be obtained as

$$\begin{vmatrix} T_{11} - \lambda & T_{12} & T_{13} \\ T_{12} & T_{22} - \lambda & T_{23} \\ T_{13} & T_{23} & T_{33} - \lambda \end{vmatrix} = (T_{11} - \lambda)(T_{22} - \lambda)(T_{33} - \lambda) + \\ T_{12}T_{23}T_{13} + T_{13}T_{12}T_{23} - T_{13}^2(T_{22} - \lambda) - T_{23}^2(T_{11} - \lambda) - (T_{33} - \lambda)T_{12}^2 = 0$$

This third order polynomial equation can be summarized as

$$\lambda^3 - I_1 \lambda^2 + I_2 \lambda - I_3 = 0 \quad (1.45)$$

where the invariants of the second-order tensor  $\mathbf{T}$  were introduced as

$$I_1 = T_{ii}, \quad I_2 = [T_{ii}T_{jj} - T_{ij}T_{ij}]/2, \quad I_3 = \det(T_{ij}) \quad (1.46)$$

After solving the three eigenvalues  $\lambda^{(1)}, \lambda^{(2)}, \lambda^{(3)}$  from (1.45) we can solve the corresponding eigenvectors  $\hat{\mathbf{n}}^{(1)}, \hat{\mathbf{n}}^{(2)}, \hat{\mathbf{n}}^{(3)}$  from (1.43).

**Problem 14** 2.30 a in Reddy

Find eigenvalues and eigenvectors of:

$$\begin{pmatrix} 4 & -4 & 0 \\ -4 & 0 & 0 \\ 0 & 0 & 3 \end{pmatrix}$$

*Answers:*

$$\lambda_1 \approx -2.47 \quad \lambda_2 = 3 \quad \lambda_3 \approx 6.47$$

$$\mathbf{e}_1^* = (-0.53, -0.85, 0)^T \quad \mathbf{e}_2^* = (0, 0, 1)^T \quad \mathbf{e}_3^* = (-0.85, 0.53, 0)^T$$

**Problem 15** 2.30 b in Reddy

Find eigenvalues and eigenvectors of:

$$\begin{pmatrix} 2 & -\sqrt{3} & 0 \\ -\sqrt{3} & 4 & 0 \\ 0 & 0 & 4 \end{pmatrix}$$

Answers:

$$\lambda_1 = 1 \quad \lambda_2 = 4 \quad \lambda_3 = 5$$

$$\mathbf{e}_1^* = \pm \left( \frac{\sqrt{3}}{2}, \frac{1}{2}, 0 \right)^T \quad \mathbf{e}_2^* = \pm (0, 0, 1)^T \quad \mathbf{e}_3^* = \pm \left( \frac{1}{2}, -\frac{\sqrt{3}}{2}, 0 \right)^T$$

**Hand-in assignment 5** a) Use the fact that eigenvectors  $\hat{\mathbf{n}}^{(i)}$  of a symmetric 2nd order tensor  $\mathbf{T}$  are orthogonal to show that  $\mathbf{T}$  can be expressed as

$$\mathbf{T} = \lambda^{(1)} \hat{\mathbf{n}}^{(1)} \hat{\mathbf{n}}^{(1)} + \lambda^{(2)} \hat{\mathbf{n}}^{(2)} \hat{\mathbf{n}}^{(2)} + \lambda^{(3)} \hat{\mathbf{n}}^{(3)} \hat{\mathbf{n}}^{(3)}$$

b) Given that the exponential function of a scalar and a 2nd order tensor are defined as:

$$\exp(\alpha) = \sum_{k=0}^{\infty} \frac{\alpha^k}{k!}, \quad \exp(\mathbf{T}) = \sum_{k=0}^{\infty} \frac{\mathbf{T}^k}{k!}.$$

Show that

$$\exp(\mathbf{T}) = \exp(\lambda^{(1)}) \hat{\mathbf{n}}^{(1)} \hat{\mathbf{n}}^{(1)} + \exp(\lambda^{(2)}) \hat{\mathbf{n}}^{(2)} \hat{\mathbf{n}}^{(2)} + \exp(\lambda^{(3)}) \hat{\mathbf{n}}^{(3)} \hat{\mathbf{n}}^{(3)}$$

and use this to compute  $\exp(\mathbf{T})$  where  $\mathbf{T}$  is represented by the the following matrix (in a  $\hat{\mathbf{e}}_i$  system)

$$[\mathbf{T}] = \begin{bmatrix} 6 & 4 & 0 \\ 4 & 3 & 0 \\ 0 & 0 & 2 \end{bmatrix}.$$

**Matlab example 4** An example of using Matlab commands for matrix definitions (for  $A$ ) and computing the eigenvalues and eigenvectors given below:

```
>> A=[1 2 3; 2 4 5; 3 5 6];
```

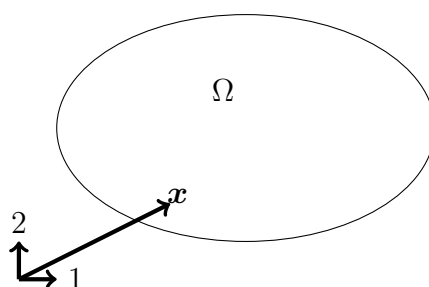
```
>> [n,lambda]=eig(A)
```

```
n =
```

0.7370	0.5910	0.3280
0.3280	-0.7370	0.5910
-0.5910	0.3280	0.7370
lambda =		
-0.5157	0	0
0	0.1709	0
0	0	11.3448

## 1.5 Spatial derivatives

For further reading see Reddy 2.4.2-2.4.3



A tensor field describes how the tensor depends on the spatial location  $\mathbf{x}$  in the body  $\Omega$  and the time  $t$ , e.g.

- Scalar field (such as temperature, pressure)  $\phi = \phi(\mathbf{x}, t)$  or  $\phi = \phi(x_i, t)$
- Vector field (such as displacement, velocity, force)  $\mathbf{u} = \mathbf{u}(\mathbf{x}, t)$  or  $u_i = u_i(x_j, t)$
- Second-order tensor field (such as stress, strain)  $\mathbf{T} = \mathbf{T}(\mathbf{x}, t)$  or  $T_{ij} = T_{ij}(x_k, t)$ .

To measure how such quantities change within the body the gradient (differential vector) operator  $\nabla$  is introduced as

$$\nabla = \hat{\mathbf{e}}_i \frac{\partial}{\partial x_i} \quad (1.47)$$

By applying the gradient operator via an open product (from the left) to a scalar field  $\phi(\mathbf{x}, t)$ , a vector field  $\mathbf{u}(\mathbf{x}, t)$  and a second-order tensor field  $\mathbf{T}(\mathbf{x}, t)$  the following results are obtained

$$\nabla \phi = \frac{\partial \phi}{\partial x_i} \hat{\mathbf{e}}_i, \quad \nabla \mathbf{u} = \frac{\partial u_j}{\partial x_i} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_j, \quad \nabla \mathbf{T} = \frac{\partial T_{jk}}{\partial x_i} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_j \hat{\mathbf{e}}_k. \quad (1.48)$$

It can be noted that the tensor fields are always increased by one degree in this using this procedure. Later we will also need to apply  $\nabla$  from the right on a vector field  $\mathbf{u}(\mathbf{x}, t)$

which is defined as

$$\mathbf{u}\nabla = \frac{\partial u_i}{\partial x_j} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_j \quad (1.49)$$

By instead applying the gradient operator via a scalar product (from the left) to a vector field  $\mathbf{u}(\mathbf{x}, t)$  and a second-order tensor field  $\mathbf{T}(\mathbf{x}, t)$  result in

$$\nabla \cdot \mathbf{u} = \frac{\partial u_i}{\partial x_i}, \quad \nabla \cdot \mathbf{T} = \frac{\partial T_{ij}}{\partial x_i} \hat{\mathbf{e}}_j. \quad (1.50)$$

This is also called the divergence with the following notation

$$\operatorname{div}(\mathbf{u}) = \nabla \cdot \mathbf{u}, \quad \operatorname{div}(\mathbf{T}) = \nabla \cdot \mathbf{T}. \quad (1.51)$$

For the divergence operator the tensor fields are always decreased by one degree.

Another product that can be used with the gradient operator is the vector product. The vector product with the gradient operator defines the curl of a vector field

$$\operatorname{curl}(\mathbf{u}) = \nabla \times \mathbf{u} = e_{ijk} \partial_i u_j \hat{\mathbf{e}}_k \quad (1.52)$$

To further compress the notation we introduce the index form of the gradient operator,  $\partial_j = \partial/\partial x_j = \hat{\mathbf{e}}_j \cdot \nabla$ , or even more compactly, a subscripted comma which for example results in:

$$\nabla \mathbf{u} = \partial_i u_j \hat{\mathbf{e}}_i \hat{\mathbf{e}}_j = u_{j,i} \hat{\mathbf{e}}_i \hat{\mathbf{e}}_j, \quad \operatorname{div}(\mathbf{T}) = \partial_i T_{ij} \hat{\mathbf{e}}_j = T_{ij,i} \hat{\mathbf{e}}_j. \quad (1.53)$$

Later in this text we will omit the base vectors and simply work with the components of the tensors e.g.  $\partial u_i/\partial x_j$ ,  $u_{i,j}$ ,  $T_{ij,j}$  etc.

**Problem 16** Show that:

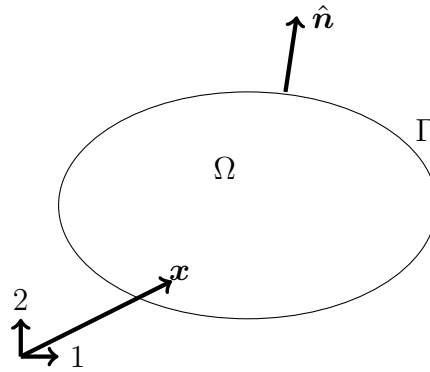
- a)  $\nabla(\mathbf{a} \cdot \mathbf{x}) = \mathbf{a} + \nabla \mathbf{a} \cdot \mathbf{x}$
- b)  $\nabla \cdot (\mathbf{a} \times \mathbf{b}) = (\nabla \times \mathbf{a}) \cdot \mathbf{b} - (\nabla \times \mathbf{b}) \cdot \mathbf{a}$
- c)  $\nabla \cdot (\mathbf{A}^T \cdot \mathbf{b}) = (\nabla \cdot \mathbf{A}) \cdot \mathbf{b} + \mathbf{A} : \nabla \mathbf{b}$

**Hand-in assignment 6** If  $\mathbf{a}$  is a vector. Show that:

- a)  $\nabla \cdot (\nabla \times \mathbf{a}) = 0$
- b)  $\mathbf{a} \times (\nabla \times \mathbf{a}) = 1/2 \nabla(\mathbf{a} \cdot \mathbf{a}) - \mathbf{a} \cdot \nabla \mathbf{a}$

## 1.6 Divergence theorem

For further reading see Reddy 2.4.5



Gauss' divergence theorem is an important and useful theorem, which allows us to convert the volume integral of a divergence into a surface integral as follows

$$\int_{\Omega} \nabla \cdot \mathbf{u} \, d\mathbf{x} = \oint_{\Gamma} \hat{\mathbf{n}} \cdot \mathbf{u} \, ds \quad \text{or} \quad \int_{\Omega} u_{i,i} \, d\mathbf{x} = \oint_{\Gamma} \hat{n}_i u_i \, ds \quad (1.54)$$

where  $\Gamma$  is the closed boundary surface of  $\Omega$ , and  $\hat{\mathbf{n}}$  is the outward normal unit vector to  $\Gamma$ . This theorem can now be applied for tensor fields  $\mathbf{T}$  by setting  $u_i = T_{i1}$ ,  $T_{i2}$  and  $T_{i3}$ . Thereby we obtain

$$\begin{cases} \int_{\Omega} T_{i1,i} \, d\mathbf{x} &= \oint_{\Gamma} \hat{n}_i T_{i1} \, ds \\ \int_{\Omega} T_{i2,i} \, d\mathbf{x} &= \oint_{\Gamma} \hat{n}_i T_{i2} \, ds \\ \int_{\Omega} T_{i3,i} \, d\mathbf{x} &= \oint_{\Gamma} \hat{n}_i T_{i3} \, ds \end{cases}$$

which can be summarized as:

$$\int_{\Omega} T_{ij,i} \, d\mathbf{x} = \oint_{\Gamma} \hat{n}_i T_{ij} \, ds \quad \text{or} \quad \int_{\Omega} \text{div}(\mathbf{T}) \, d\mathbf{x} = \oint_{\Gamma} \hat{\mathbf{n}} \cdot \mathbf{T} \, ds \quad (1.55)$$

If we instead apply the Gauss' divergence theorem to a scalar field for the three cases

$$\begin{cases} u_1 = \phi, u_2 = u_3 = 0 \\ u_1 = 0, u_2 = \phi, u_3 = 0 \\ u_1 = 0, u_2 = 0, u_3 = \phi \end{cases}$$

we obtain

$$\int_{\Omega} \partial \phi_{,i} \, d\mathbf{x} = \oint_{\Gamma} \hat{n}_i \phi \, ds \quad \text{or} \quad \int_{\Omega} \nabla \phi \, d\mathbf{x} = \oint_{\Gamma} \hat{\mathbf{n}} \phi \, ds \quad (1.56)$$

In practice, the name divergence theorem refers to equations (1.54), (1.55) and (1.56).

**Problem 17** By using the divergence theorem show that:  $\oint_{\Gamma} x_i \hat{n}_j \, ds = V \delta_{ij}$ .

**Hand-in assignment 7** Prove the following formula:  $\int_{\Omega} \varphi_i \partial_j \sigma_{ij} \, d\mathbf{x} = \oint_{\Gamma} \hat{n}_j \sigma_{ij} \varphi_i \, ds - \int_{\Omega} \sigma_{ij} \partial_j \varphi_i \, d\mathbf{x}$ .





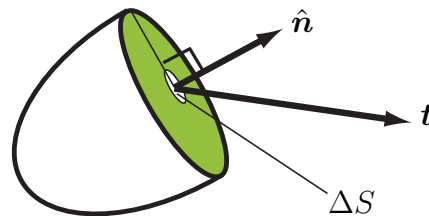
## 2. Stress, motion and deformation

### 2.1 Stress analysis

For further reading see Reddy 4.1-4.3.2

The stress (also called traction) vector  $\mathbf{t}(\hat{\mathbf{n}})$  is defined as the force acting on an area with normal  $\hat{\mathbf{n}}$ . In a point of a body the stress vector is defined as

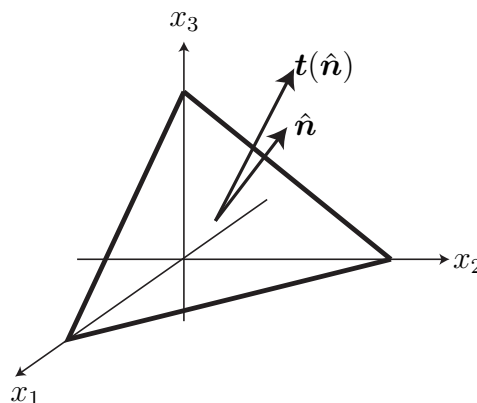
$$\mathbf{t}(\hat{\mathbf{n}}) = \lim_{\Delta S \rightarrow 0} \frac{\Delta \mathbf{F}}{\Delta S} \quad (2.1)$$



A property of the stress vector is that it must follow Newton's third law for action and reaction. Therefore, in the same point of a body the stress vector on the area with normal  $\hat{\mathbf{n}}$  and normal  $-\hat{\mathbf{n}}$  must be opposite. This means that

$$\mathbf{t}(\hat{\mathbf{n}}) = -\mathbf{t}(-\hat{\mathbf{n}}). \quad (2.2)$$

To find a relation between the normal  $\hat{\mathbf{n}}$  and the stress vector  $\mathbf{t}(\hat{\mathbf{n}})$  we study a tetrahedral element:



The tetrahedron is assumed to have the four surfaces defined as

1. normal  $\hat{\mathbf{n}}$  and area  $A$  subjected to stress vector  $\mathbf{t}(\hat{\mathbf{n}})$ ,
2. normal  $-\hat{\mathbf{e}}_1$  and area  $A_1$  subjected to stress vector  $\mathbf{t}(-\hat{\mathbf{e}}_1)$ ,
3. normal  $-\hat{\mathbf{e}}_2$  and area  $A_2$  subjected to stress vector  $\mathbf{t}(-\hat{\mathbf{e}}_2)$ ,
4. normal  $-\hat{\mathbf{e}}_3$  and area  $A_3$  subjected to stress vector  $\mathbf{t}(-\hat{\mathbf{e}}_3)$ .

The relation between the areas  $A, A_1, A_2, A_3$  and the components of  $\hat{\mathbf{n}}$  can be conveniently derived by using the divergence theorem (show this at home!):

$$-\hat{\mathbf{e}}_1 A_1 - \hat{\mathbf{e}}_2 A_2 - \hat{\mathbf{e}}_3 A_3 + \hat{\mathbf{n}} A = 0. \quad (2.3)$$

From these three equations we can identify that  $A_i = \hat{n}_i A$ . Next step is now to study equilibrium of the tetrahedron:

$$\mathbf{t}(\hat{\mathbf{n}}) A + \mathbf{t}(-\hat{\mathbf{e}}_1) A_1 + \mathbf{t}(-\hat{\mathbf{e}}_2) A_2 + \mathbf{t}(-\hat{\mathbf{e}}_3) A_3 = \mathbf{0}.$$

If we use the relation between the areas and Newton's third law we obtain:

$$\mathbf{t}(\hat{\mathbf{n}}) = \mathbf{t}(\hat{\mathbf{e}}_1)\hat{n}_1 + \mathbf{t}(\hat{\mathbf{e}}_2)\hat{n}_2 + \mathbf{t}(\hat{\mathbf{e}}_3)\hat{n}_3. \quad (2.4)$$

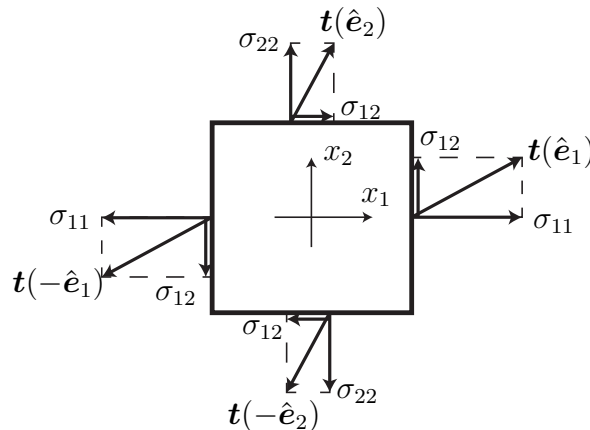
The second-order stress tensor  $\boldsymbol{\sigma}$  is defined based on  $\mathbf{t}(\hat{\mathbf{e}}_i)$  such that

$$[\sigma_{ij}] = [t_j(\hat{\mathbf{e}}_i)] \quad (2.5)$$

or more explicitly

$$\begin{bmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & \sigma_{22} & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & \sigma_{33} \end{bmatrix} = \begin{bmatrix} t_1(\hat{\mathbf{e}}_1) & t_2(\hat{\mathbf{e}}_1) & t_3(\hat{\mathbf{e}}_1) \\ t_1(\hat{\mathbf{e}}_2) & t_2(\hat{\mathbf{e}}_2) & t_3(\hat{\mathbf{e}}_2) \\ t_1(\hat{\mathbf{e}}_3) & t_2(\hat{\mathbf{e}}_3) & t_3(\hat{\mathbf{e}}_3) \end{bmatrix} \quad (2.6)$$

This can be graphically shown as (here 2d):



To sum up, the relation between the stress vector  $\mathbf{t}$  and normal vector  $\hat{\mathbf{n}}$  is obtained via the stress tensor  $\boldsymbol{\sigma}$  as follows:

$$\mathbf{t} = \mathbf{n} \cdot \boldsymbol{\sigma} = \boldsymbol{\sigma}^T \cdot \mathbf{n} \quad \text{or } t_i = n_j \sigma_{ji} = \sigma_{ij}^T n_j. \quad (2.7)$$

This relation is the so-called Cauchy's formula. As will be proven later in the course, the stress tensor is symmetric due to principle of angular momentum i.e.  $\boldsymbol{\sigma} = \boldsymbol{\sigma}^T$  and, hence, this relation can be written as  $\mathbf{t} = \boldsymbol{\sigma} \cdot \hat{\mathbf{n}}$ . We can conclude that if the stress tensor  $\boldsymbol{\sigma}$  is known in a point of the body then it is possible to compute the stress vector  $\mathbf{t}$  on any plane through the point. This is called the Cauchy's stress principle.

Often the components of the stress tensor are divided into normal stresses and shear stresses. The normal stresses are the diagonal components of the stress tensor i.e.  $\sigma_{11}$ ,  $\sigma_{22}$  and  $\sigma_{33}$  whereas the shear stresses are the off-diagonal components i.e.  $\sigma_{12}$ ,  $\sigma_{23}$  and  $\sigma_{13}$ . Note that the terminology normal and shear components relate to what plane that is chosen. In the figure above the choice of plane is defined by the normal  $\hat{\mathbf{e}}_1$  or  $\hat{\mathbf{e}}_2$ . In general, the normal component of the stress on a plane with normal  $\hat{\mathbf{n}}$  is obtained from

$$\sigma_{nn} = \hat{\mathbf{n}} \cdot \mathbf{t} = \hat{\mathbf{n}} \cdot \boldsymbol{\sigma} \cdot \hat{\mathbf{n}} = \boldsymbol{\sigma} : (\hat{\mathbf{n}} \hat{\mathbf{n}}) = \sigma_{ij} \hat{n}_i \hat{n}_j. \quad (2.8)$$

Let us now adopt the concept of eigenvalues and eigenvectors for a stress tensor  $\boldsymbol{\sigma}$ . The eigenvector is a direction  $\hat{\mathbf{n}}$  that is not changed upon a scalar multiplication with the stress tensor  $\boldsymbol{\sigma}$ :

$$\hat{\mathbf{n}} \xrightarrow{\boldsymbol{\sigma}} \mathbf{t} = \lambda \hat{\mathbf{n}}$$

This means that on a plane with the normal being an eigenvector of  $\boldsymbol{\sigma}$  then the stress vector  $\mathbf{t}$  is parallel to the normal i.e.  $\mathbf{t} = \lambda \hat{\mathbf{n}}$ . In other words, on such a plane only the normal components are non-zero.

Often the stress tensor  $\boldsymbol{\sigma}$  is additively decomposed into a deviatoric  $\boldsymbol{\sigma}'$  and a spherical (hydrostatic) tensor  $\sigma_m \boldsymbol{\delta}$  as follows:

$$\boldsymbol{\sigma} = \boldsymbol{\sigma}' + \sigma_m \boldsymbol{\delta} \quad \text{or } \sigma_{ij} = \sigma'_{ij} + \sigma_m \delta_{ij} \quad (2.9)$$

with

$$\sigma_m = \sigma_{kk}/3 \quad \text{and } \sigma'_{ij} = \sigma_{ij} - \sigma_m \delta_{ij}. \quad (2.10)$$

**Problem 18** 4.1 in Reddy

**4.1** Suppose that  $\mathbf{t}^{\hat{\mathbf{n}}_1}$  and  $\mathbf{t}^{\hat{\mathbf{n}}_2}$  are stress vectors acting on planes with unit normals  $\hat{\mathbf{n}}_1$  and  $\hat{\mathbf{n}}_2$ , respectively, and passing through a point with the stress state  $\sigma$ . Show that the component of  $\mathbf{t}^{\hat{\mathbf{n}}_1}$  along  $\hat{\mathbf{n}}_2$  is equal to the component of  $\mathbf{t}^{\hat{\mathbf{n}}_2}$  along the normal  $\mathbf{t}^{\hat{\mathbf{n}}_1}$ .

**Problem 19** 4.4 a,b,c in Reddy

**4.4** Consider a kinematically infinitesimal stress field whose matrix of scalar components in the vector basis  $\{\hat{\mathbf{e}}_i\}$  is

$$\begin{bmatrix} 1 & 0 & 2X_2 \\ 0 & 1 & 4X_1 \\ 2X_2 & 4X_1 & 1 \end{bmatrix} \times 10^3 \text{ (psi)},$$

where the Cartesian coordinate variables  $X_i$  are in inches (in.) and the units of stress are pounds per square inch (psi).

- Determine the traction vector acting at point  $\mathbf{X} = \hat{\mathbf{e}}_1 + 2\hat{\mathbf{e}}_2 + 3\hat{\mathbf{e}}_3$  on the plane  $X_1 + X_2 + X_3 = 6$ .
- Determine the normal and projected shear tractions acting at this point on this plane.
- Determine the principal stresses and principal directions of stress at this point.

*Answers:*

a)

$$\mathbf{t}_1 = \frac{1}{\sqrt{3}} (5\mathbf{e}_1 + 5\mathbf{e}_2 + 9\mathbf{e}_3)^T$$

b)

$$t_{nn} = 19/3 \cdot 10^3 \text{ psi} \quad t_{ns} = \sqrt{32}/3 \cdot 10^3 \text{ psi}$$

c)

$$\lambda_1 = 1 - 4\sqrt{2} \quad \lambda_2 = 1 \quad \lambda_3 = 1 + 4\sqrt{2}$$

$$\mathbf{e}_1^* = \frac{1}{2} (1, 1, -\sqrt{2})^T \quad \mathbf{e}_2^* = \frac{1}{2} (\sqrt{2}, -\sqrt{2}, 0)^T \quad \mathbf{e}_3^* = \frac{1}{2} (1, 1, \sqrt{2})^T$$

**Problem 20** 4.6 in Reddy

**4.6** For the state of stress given in Problem 4.5, determine the normal and shear stresses on a plane intersecting the point where the plane is defined by the points  $(0, 0, 0)$ ,  $(2, -1, 3)$ , and  $(-2, 0, 1)$ .

**4.5** The three-dimensional state of stress at a point  $(1, 1, -2)$  within a body relative to the coordinate system  $(x_1, x_2, x_3)$  is

$$\begin{bmatrix} 2.0 & 3.5 & 2.5 \\ 3.5 & 0.0 & -1.5 \\ 2.5 & -1.5 & 1.0 \end{bmatrix} \times 10^6 \text{ (Pa)}.$$

*Answers:* (two possible directions of the normal)

$$t_{nn} = \pm 0.3478 \text{ MPa} \quad t_{ns} = \pm 4.2955 \text{ MPa}$$

**Problem 21** 4.12 in Reddy

**4.12** Find the values of  $\sigma_s$  and  $\sigma_{22}$  for the state of stress shown in Figure P4.12.

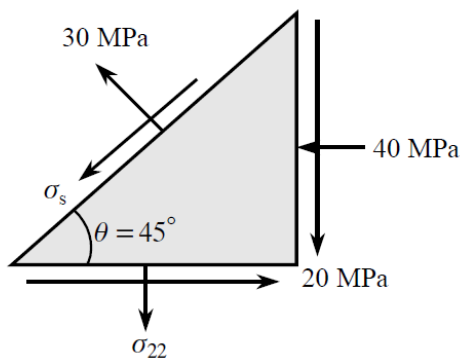


Figure P4.12.

*Answers:*

$$\sigma_s = -50 \text{ MPa} \quad \sigma_{22} = 60 \text{ MPa}$$

**Problem 22** 4.18 in Reddy

**4.18** Given the following state of stress ( $\sigma_{ij} = \sigma_{ji}$ ),

$$\sigma_{11} = -2x_1^2, \quad \sigma_{12} = -7 + 4x_1x_2 + x_3, \quad \sigma_{13} = 1 + x_1 - 3x_2,$$

$$\sigma_{22} = 3x_1^2 - 2x_2^2 + 5x_3, \quad \sigma_{23} = 0, \quad \sigma_{33} = -5 + x_1 + 3x_2 + 3x_3.$$

determine (a) the stress vector at point  $(x_1, x_2, x_3)$  on the plane  $x_1 + x_2 + x_3 = \text{constant}$ , (b) the normal and shearing components of the stress vector at point  $(1, 1, 3)$ , and (c) the principal stresses and their orientation at point  $(1, 2, 1)$ .

*Answers:*

a)

$$t_1 = \frac{1}{\sqrt{3}}(-2x_1^2 - 6 + 4x_1x_2 + x_3 + x_1 - 3x_2)$$

$$t_2 = \frac{1}{\sqrt{3}}(-7 + 4x_1x_2 + 3x_1^2 - 2x_2^2 + 6x_3)$$

$$t_3 = \frac{1}{\sqrt{3}}(-4 + 2x_1 + 3x_3)$$

b)

$$t_{nn} = 20/3 \quad t_{ns} = \sqrt{542/9}$$

c)

$$\lambda_1 \approx -4.5530 \quad \lambda_2 \approx 0.6349 \quad \lambda_3 \approx 6.9184$$

$$\mathbf{e}_1^* \approx (-0.8549, 0.3755, -0.3580)^T$$

$$\mathbf{e}_2^* \approx (0.2916, 0.9185, 0.2672)^T$$

$$\mathbf{e}_3^* \approx (-0.4291, -0.1240, 0.8947)^T$$

**Hand-in assignment 8** Given the stress tensor  $\boldsymbol{\sigma}$  (here represented in a matrix format with components in a  $\hat{\mathbf{e}}_1, \hat{\mathbf{e}}_2, \hat{\mathbf{e}}_3$ -system)

$$[\boldsymbol{\sigma}] = \begin{bmatrix} 30 & 0 & 10 \\ 0 & 30 & 10 \\ 10 & 10 & 30 \end{bmatrix} \text{ MPa}$$

Compute the corresponding deviatoric stress tensor  $\boldsymbol{\sigma}'$ . For the deviatoric stress compute the principal stresses, principal directions, invariants (acc to (1.46)) and the obtained stress vector on the plane with normal

$$\hat{\mathbf{n}} = \frac{1}{\sqrt{5}}(\hat{\mathbf{e}}_1 + 2\hat{\mathbf{e}}_2).$$

## 2.2 Continuum motion

For further reading see Reddy 3.1-3.3.1, 3.4.2, 5.2.2

The motion of a continuum (material volume) is shown in figure 2.1. A material particle P may be identified by its initial (or reference) position  $\mathbf{X}$ . The current position  $\mathbf{x}$ , of a material particle is then defined by a function

$$x_i = x_i(\mathbf{X}, t) \quad (2.11)$$

The displacement  $\mathbf{u}$  of a particle P is defined as

$$u_i = x_i - X_i \quad (2.12)$$

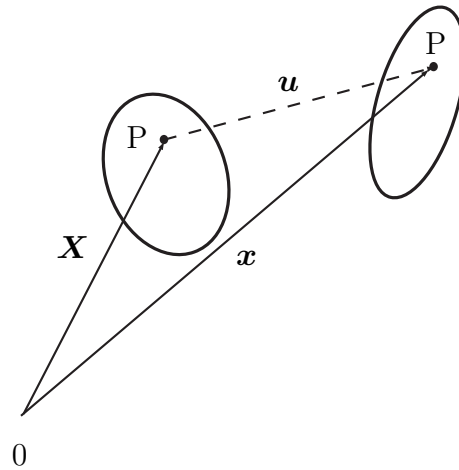


Figure 2.1: Illustration of motion of a continuum.

A key quantity that describes the deformation of the body (material volume) is the *deformation gradient*  $\mathbf{F}$ . The deformation gradient describes the relation between a line element  $d\mathbf{X}$  at the material particle  $P$  in the initial (undeformed) body and the corresponding line element  $d\mathbf{x}$  at the material particle  $P$  in the current (undeformed) body, i.e.

$$d\mathbf{x} = \mathbf{F} \cdot d\mathbf{X} \quad \text{or} \quad F_{ij} = \frac{\partial x_i}{\partial X_j} = \delta_{ij} + \frac{\partial u_i}{\partial X_j} \quad \text{or} \quad \mathbf{F} = \mathbf{x}\nabla_0 \quad (2.13)$$

which is also illustrated in figure 2.2. Based on the deformation gradient  $\mathbf{F}$  a number of

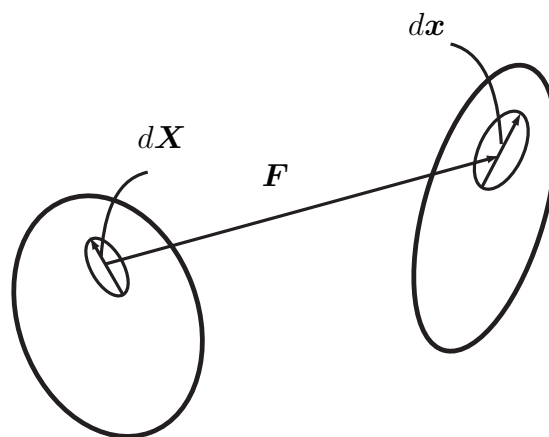


Figure 2.2: Illustration of deformation gradient.

strain measures can be defined. An example is the frequently used Green-Lagrange strain

$\mathbf{E}$  defined as follows:

$$\mathbf{E} = \frac{1}{2} (\mathbf{F}^T \cdot \mathbf{F} - \boldsymbol{\delta}) \quad \text{or} \quad (2.14)$$

$$E_{ij} = \frac{1}{2} (F_{ik}^T F_{kj} - \delta_{ij}) = \frac{1}{2} \left( \frac{\partial u_i}{\partial X_j} + \frac{\partial u_j}{\partial X_i} + \frac{\partial u_k}{\partial X_i} \frac{\partial u_k}{\partial X_j} \right) \quad (2.15)$$

For the special case of small deformations  $\mathbf{E}$  approaches the usual small strain tensor  $\boldsymbol{\epsilon}$ , i.e.

$$E_{ij} \approx \frac{1}{2} \left( \frac{\partial u_i}{\partial X_j} + \frac{\partial u_j}{\partial X_i} \right) \approx \frac{1}{2} \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) = \epsilon_{ij} \quad (2.16)$$

We can note that both the Green-Lagrange strain  $\mathbf{E}$  and the small strain tensor  $\boldsymbol{\epsilon}$  are symmetric, i.e.  $\mathbf{E}^T = \mathbf{E}$  and  $\boldsymbol{\epsilon}^T = \boldsymbol{\epsilon}$ .

## 2.3 Lagrangian and Eulerian description

Physical field quantities can be described in either a *Lagrangian* (or sometimes called material) or *Eulerian* description:

- Lagrangian description of scalars, vectors and second-order tensors:

$$\begin{aligned} \phi &= \phi(\mathbf{X}, t), \quad \mathbf{u} = \mathbf{u}(\mathbf{X}, t), \quad \mathbf{T} = \mathbf{T}(\mathbf{X}, t) \quad \text{or} \\ \phi &= \phi(X_i, t), \quad u_i = u_i(X_j, t), \quad T_{ij} = T_{ij}(X_k, t) \end{aligned}$$

- Eulerian description of scalars, vectors and second-order tensors:

$$\begin{aligned} \phi &= \phi(\mathbf{x}, t), \quad \mathbf{u} = \mathbf{u}(\mathbf{x}, t), \quad \mathbf{T} = \mathbf{T}(\mathbf{x}, t) \quad \text{or} \\ \phi &= \phi(x_i, t), \quad u_i = u_i(x_j, t), \quad T_{ij} = T_{ij}(x_k, t) \end{aligned}$$

An important field quantity is the velocity  $\mathbf{v}$  of a material particle P. The velocity is defined as the time derivative of the position vector  $\mathbf{x}$ :

$$\mathbf{v} = \frac{d\mathbf{x}(\mathbf{X}, t)}{dt} \quad \text{or} \quad v_i = \frac{dx_i(X_j, t)}{dt} \quad (2.17)$$

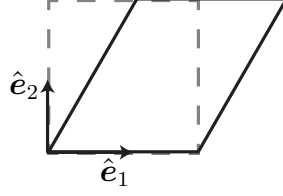
whereby the velocity is described in an Lagrangian description  $\mathbf{v}(\mathbf{X}, t)$ . By assuming that the initial position of the particle  $\mathbf{X}$  can be expressed in terms of  $\mathbf{x}$  and  $t$  we can write the velocity in Eulerian description

$$\mathbf{v} = \mathbf{v}(\mathbf{x}, t) \quad (2.18)$$

Next follows three examples to illustrate the introduced concepts regarding motion.



■ **Example 2.1** Simple shear of a quadratic disc (side length  $h_0$ ) where the upper boundary moves horizontally with velocity  $v_0$ :



The motion can be expressed as:

$$\begin{cases} x_1(X_1, X_2, X_3, t) = X_1 + X_2 v_0 t / h_0 \\ x_2(X_1, X_2, X_3, t) = X_2 \\ x_3(X_1, X_2, X_3, t) = X_3 \end{cases}$$

whereby the velocity  $\mathbf{v}$  can be obtained, in Lagrangian description, as:

$$v_i = \begin{bmatrix} X_2 v_0 / h_0 \\ 0 \\ 0 \end{bmatrix}$$

By using the expression for the motion the velocity can be written in an Eulerian description as

$$v_i = \begin{bmatrix} x_2 v_0 / h_0 \\ 0 \\ 0 \end{bmatrix}$$

Based on the expression for the motion we can also obtain the deformation gradient  $\mathbf{F}$  as

$$F_{ij} = \begin{bmatrix} 1 & v_0 t / h_0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

and the Green Lagrange strain  $\mathbf{E}$

$$E_{ij} = \frac{1}{2} (F_{ik}^T F_{kj} - \delta_{ij}) = \dots = \frac{1}{2} \begin{bmatrix} 0 & v_0 t / h_0 & 0 \\ v_0 t / h_0 & (v_0 t / h_0)^2 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

The displacement vector  $\mathbf{u} = \mathbf{x} - \mathbf{X}$  is given as:

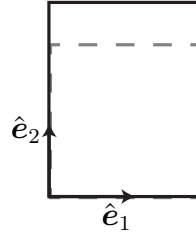
$$u_i = \begin{bmatrix} X_2 v_0 t / h_0 \\ 0 \\ 0 \end{bmatrix}$$

whereby the small strain tensor  $\epsilon$  becomes

$$\epsilon_{ij} = \frac{1}{2} \left( \frac{\partial u_i}{\partial X_j} + \frac{\partial u_j}{\partial X_i} \right) = \begin{bmatrix} 0 & (v_0 t/h_0)/2 & 0 \\ (v_0 t/h_0)/2 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

■

■ **Example 2.2** Pure elongation of a quadratic disc (side length  $h_0$ ) where the upper boundary moves vertically with velocity  $v_0$ :



The motion can be expressed as:

$$\begin{cases} x_1(X_1, X_2, X_3, t) = X_1 \\ x_2(X_1, X_2, X_3, t) = X_2 + X_2 v_0 t/h_0 \\ x_3(X_1, X_2, X_3, t) = X_3 \end{cases}$$

whereby the velocity  $\mathbf{v}$  can be obtained, in Lagrangian description, as:

$$v_i = \begin{bmatrix} 0 \\ X_2 v_0/h_0 \\ 0 \end{bmatrix}$$

By using the expression for the motion the velocity can be written in an Eulerian description as

$$v_i = \begin{bmatrix} 0 \\ x_2 v_0/(h_0 + v_0 t) \\ 0 \end{bmatrix}$$

Based on the expression for the motion we can also obtain the deformation gradient  $\mathbf{F}$  as

$$F_{ij} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 + v_0 t/h_0 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

and the Green Lagrange strain  $\mathbf{E}$

$$E_{ij} = \frac{1}{2} \left( F_{ik}^T F_{kj} - \delta_{ij} \right) = \dots = \begin{bmatrix} 0 & 0 & 0 \\ 0 & (v_0 t/h_0)^2/2 + v_0 t/h_0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

The displacement vector  $\mathbf{u} = \mathbf{x} - \mathbf{X}$  is given as:

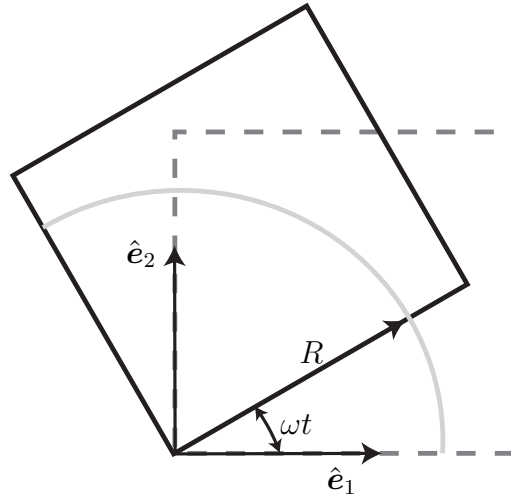
$$u_i = \begin{bmatrix} 0 \\ X_2 v_0 t / h_0 \\ 0 \end{bmatrix}$$

whereby the small strain tensor  $\boldsymbol{\epsilon}$  becomes

$$\epsilon_{ij} = \frac{1}{2} \left( \frac{\partial u_i}{\partial X_j} + \frac{\partial u_j}{\partial X_i} \right) = \begin{bmatrix} 0 & 0 & 0 \\ 0 & v_0 t / h_0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

■

■ **Example 2.3** Pure rotation around the left corner of a quadratic disc (side length  $h_0$ ) with rotational velocity  $\omega$ :



An arbitrary point's initial location in the disc is described by the distance  $R = \sqrt{X_1^2 + X_2^2}$  to the left corner and angle  $\alpha_0 = \text{atan}(X_2/X_1)$  (from the  $\hat{e}_1$  axis). During rotation the angle changes with rotation according to  $\alpha = \alpha_0 + \omega t$  whereby the motion can be expressed as:

$$\begin{cases} x_1(X_1, X_2, X_3, t) = R \cos(\alpha) \\ x_2(X_1, X_2, X_3, t) = R \sin(\alpha) \\ x_3(X_1, X_2, X_3, t) = X_3 \end{cases}$$

which can be (after some manipulations) written as

$$\begin{cases} x_1(X_1, X_2, X_3, t) = X_1 \cos(\omega t) - X_2 \sin(\omega t) \\ x_2(X_1, X_2, X_3, t) = X_1 \sin(\omega t) + X_2 \cos(\omega t) \\ x_3(X_1, X_2, X_3, t) = X_3 \end{cases}$$

The velocity  $\mathbf{v}$  can be obtained, in Lagrangian and Eulerian description, as :

$$v_i = \begin{bmatrix} \omega (-X_1 \sin(\omega t) - X_2 \cos(\omega t)) \\ \omega (X_1 \cos(\omega t) - X_2 \sin(\omega t)) \\ 0 \end{bmatrix} = \omega \begin{bmatrix} -x_2 \\ x_1 \\ 0 \end{bmatrix}$$

Based on the expression for the motion we can also obtain the deformation gradient  $\mathbf{F}$  as

$$F_{ij} = \begin{bmatrix} \cos(\omega t) & -\sin(\omega t) & 0 \\ \sin(\omega t) & \cos(\omega t) & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

and the Green Lagrange strain  $\mathbf{E}$

$$E_{ij} = \frac{1}{2} (F_{ik}^T F_{kj} - \delta_{ij}) = \dots = \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

The displacement vector  $\mathbf{u} = \mathbf{x} - \mathbf{X}$  is given as:

$$u_i = \begin{bmatrix} X_1 (\cos(\omega t) - 1) - X_2 \sin(\omega t) \\ X_1 \sin(\omega t) + X_2 (\cos(\omega t) - 1) \\ 0 \end{bmatrix}$$

whereby the small strain tensor  $\boldsymbol{\epsilon}$  becomes

$$\epsilon_{ij} = \frac{1}{2} \left( \frac{\partial u_i}{\partial X_j} + \frac{\partial u_j}{\partial X_i} \right) = \begin{bmatrix} \cos(\omega t) - 1 & 0 & 0 \\ 0 & \cos(\omega t) - 1 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

■

**Problem 23** 3.3 in Reddy

**3.3** The motion of a body is described by the mapping

$$\chi(\mathbf{X}) = (X_1 + t^2 X_2) \hat{\mathbf{e}}_1 + (X_2 + t^2 X_1) \hat{\mathbf{e}}_2 + X_3 \mathbf{e}_3,$$

where  $t$  denotes time. Determine

- the components of the deformation gradient tensor  $\mathbf{F}$ ,
- the components of the displacement, velocity, and acceleration vectors, and
- the position  $(X_1, X_2, X_3)$  of the particle in undeformed configuration that occupies the position  $(x_1, x_2, x_3) = (9, 6, 1)$  at time  $t = 2$  s in the deformed configuration.

*Answers:*

a)

$$\mathbf{F} = \begin{pmatrix} 1 & t^2 & 0 \\ t^2 & 1 & 0 \\ 0 & 0 & 1 \end{pmatrix}$$

b)

$$\mathbf{u} = (t^2 X_2) \mathbf{e}_1 + (t^2 X_1) \mathbf{e}_2$$

$$\mathbf{v} = (2t X_2) \mathbf{e}_1 + (2t X_1) \mathbf{e}_2$$

$$\mathbf{a} = (2X_2) \mathbf{e}_1 + (2X_1) \mathbf{e}_2$$

c)

$$\mathbf{X} = \mathbf{e}_1 + 2\mathbf{e}_2 + \mathbf{e}_3$$

**Problem 24** 3.6 a,b in Reddy**3.6** Suppose that the motion of a continuous medium is given by

$$x_1 = X_1 \cos At + X_2 \sin At,$$

$$x_2 = -X_1 \sin At + X_2 \cos At,$$

$$x_3 = (1 + Bt)X_3,$$

where  $A$  and  $B$  are constants. Determine the components of

(a) the displacement vector in the material description,

(b) the displacement vector in the spatial description

*Answers:*

a)

$$u_1(\mathbf{X}) = X_1 \cos At + X_2 \sin At - X_1$$

$$u_2(\mathbf{X}) = -X_1 \sin At + X_2 \cos At - X_2$$

$$u_3(\mathbf{X}) = BtX_3$$

b)

$$u_1(\mathbf{x}) = x_1 - x_1 \cos At + x_2 \sin At$$

$$u_2(\mathbf{x}) = x_2 - x_1 \sin At + x_2 \cos At$$

$$u_3(\mathbf{x}) = \frac{Bt}{1+Bt} x_3$$

**Problem 25** 3.7 in Reddy (not Eulerian)

**3.7** If the deformation mapping of a body is given by

$$\chi(\mathbf{X}) = (X_1 + AX_2)\hat{\mathbf{e}}_1 + (X_2 + BX_1)\hat{\mathbf{e}}_2 + X_3\hat{\mathbf{e}}_3,$$

where  $A$  and  $B$  are constants, determine

- the displacement components in the material description,
- the displacement components in the spatial description, and
- the components of the Green–Lagrange strain tensor

*Answers:*

a)

$$u(\mathbf{X}) = (AX_2)\mathbf{e}_1 + (BX_1)\mathbf{e}_2$$

b)

$$u(\mathbf{x}) = \frac{A}{1-AB}(x_2 - Bx_1)\mathbf{e}_1 + \frac{B}{1-AB}(x_1 - Ax_2)\mathbf{e}_2$$

c)

$$\mathbf{E} = \frac{1}{2} \begin{pmatrix} B^2 & A+B & 0 \\ A+B & A^2 & 0 \\ 0 & 0 & 0 \end{pmatrix}$$

**Hand-in assignment 9** For a quadratic disc with side lengths  $h_0$  we have the motion:

$$\begin{cases} x_1(X_1, X_2, X_3, t) = X_1 + X_2 v_0 t/h_0 \\ x_2(X_1, X_2, X_3, t) = X_2 + X_1 v_0 t/h_0 \\ x_3(X_1, X_2, X_3, t) = X_3 \end{cases}$$

where  $v_0$  is a velocity.

- Illustrate how the disc will deform (in 2D).
- Determine the velocity vector in Lagrangian and Eulerian description.
- Determine the Green-Lagrange strain  $\mathbf{E}$ .

## 2.4 Material time derivative

Physical field quantities such as temperature, velocity, stress tensor change with time. This change is naturally described as the time derivative of the physical quantity of a material particle in the continuum. The particle is uniquely identified by the Lagrangian (material) vector  $\mathbf{X}$ . Therefore, it is useful to introduce the material time derivative which is denoted  $D(\bullet)/Dt$ ,  $(\dot{\bullet})$  or  $d(\bullet)/dt$ . If the physical quantity  $\phi$  is described in an

Lagrangian description  $\phi(\mathbf{X}, t)$

$$\frac{D\phi(\mathbf{X}, t)}{Dt} = \dot{\phi}(\mathbf{X}, t) = \frac{d\phi(\mathbf{X}, t)}{dt}, \quad (2.19)$$

whereas the material time derivative of a field quantity described in an Eulerian description  $\phi(\mathbf{x}, t)$  is obtained by the chain rule:

$$\begin{aligned} \dot{\phi}(\mathbf{x}(\mathbf{X}, t), t) &= \frac{d\phi(\mathbf{x}(\mathbf{X}, t), t)}{dt} \\ &= \frac{\partial\phi(\mathbf{x}, t)}{\partial x_i} \frac{\partial x_i(\mathbf{X}, t)}{\partial t} + \frac{\partial\phi(\mathbf{x}, t)}{\partial t} \Big|_{\mathbf{x}} \end{aligned} \quad (2.20)$$

The first part in the result is the convective part while the second part is the time derivative of  $\phi$  in a spatial position  $\mathbf{x}$ .

## 2.5 Reynolds' transport theorem for a material volume

In the balance laws physical quantities are integrated over the volume of interest. The integration can be performed for the current volume of the continuum  $\Omega$ :

$$\int_{\Omega} \phi \, d\mathbf{x}$$

By substituting the volume  $\Omega$  to the initial volume (undeformed) of the continuum  $\Omega_0$  we obtain (using results from Math see e.g [https://en.wikipedia.org/wiki/Integration\\_by\\_substitution](https://en.wikipedia.org/wiki/Integration_by_substitution)):

$$\int_{\Omega} \phi \, d\mathbf{x} = \int_{\Omega_0} \phi J \, d\mathbf{X} \quad (2.21)$$

where  $J = \det(\mathbf{F})$ .

*Note:* The volume change of a body  $V/V_0$  is given by  $J = \det(\mathbf{F})$ . This follows immediately from (2.21) by setting  $\phi = 1$ .

The material time derivative of a volume integral of  $\phi$  can now be obtained as (using that  $\Omega_0$  is constant):

$$\frac{d}{dt} \int_{\Omega} \phi \, d\mathbf{x} = \frac{d}{dt} \int_{\Omega_0} \phi J \, d\mathbf{X} = \int_{\Omega_0} \dot{\phi} J + \phi \dot{J} \, d\mathbf{X} \quad (2.22)$$

The time derivative of the volume change  $\dot{J}$  is given by (here without any proof):

$$\dot{J} = \operatorname{div}(\mathbf{v}) J \quad \text{or} \quad \dot{J} = v_{i,i} J \quad (2.23)$$

whereby we can obtain Reynold's transport theorem:

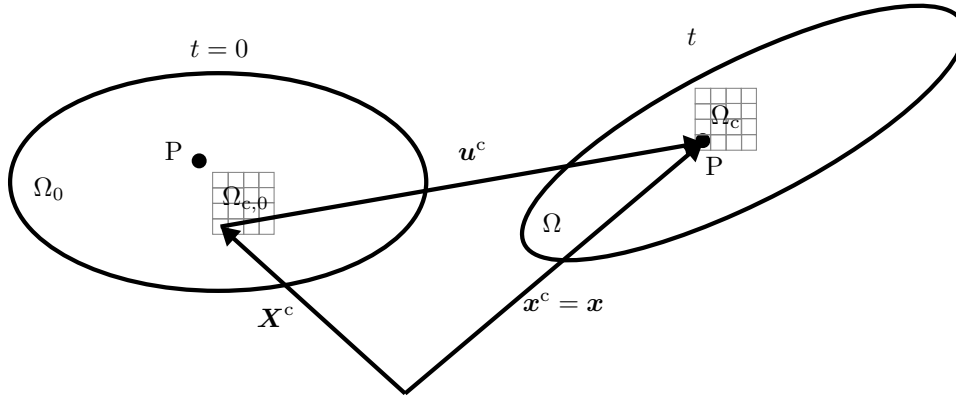
$$\frac{d}{dt} \int_{\Omega} \phi \, d\mathbf{x} = \int_{\Omega} \left( \frac{d\phi}{dt} + \phi \frac{\partial v_i}{\partial x_i} \right) d\mathbf{x}. \quad (2.24)$$

**Hand-in assignment 10** Show that (2.24) for  $\phi(x_i, t)$  can be written as:

$$\frac{d}{dt} \int_{\Omega} \phi \, d\mathbf{x} = \int_{\Omega} \frac{\partial \phi}{\partial t} \Big|_{\mathbf{x}} \, d\mathbf{x} + \oint_{\Gamma} \hat{n}_i v_i \phi \, ds$$

## 2.6 Reynolds' transport theorem for a control volume

In fluid mechanics quantities are often measured in a control volume  $\Omega_c$  with boundary  $\Gamma_c$ . This control volume do not in general follow the movement of the material particles in the continuum as is illustrated in the figure below.



At time  $t$  the position of a control volume element is  $\mathbf{x}^c$  and its velocity  $\mathbf{v}^c$ . Note that although the position of a control volume element and a material point element  $P$  coincides at time  $t$ , i.e.  $\mathbf{x}^c = \mathbf{x}$ , their velocities  $\mathbf{v}^c$  and  $\mathbf{v}$  differ.

How  $\int_{\Omega_c} \phi \, d\mathbf{x}$  changes with time can be obtained from the general form of Reynold's transport theorem:

$$\frac{d}{dt} \int_{\Omega_c} \phi \, d\mathbf{x} = \int_{\Omega_c} \left( \frac{d\phi}{dt} + \phi \frac{\partial v_i^c}{\partial x_i^c} \right) d\mathbf{x}. \quad (2.25)$$

This follows from the same arguments as in (2.22)-(2.24). Two special cases are common:

- The control volume elements are equal material point elements (they follow the deformation) then  $\mathbf{v}^c = \mathbf{v}$  and (2.24) is re-obtained. This assumption that the control volume elements are "nailed" to the material elements are often used in solid mechanics.
- The control volume elements are fixed in time which means that  $\mathbf{v}^c = \mathbf{0}$ . This gives that if  $\phi(x_i^c, t)$  we obtain:

$$\frac{d}{dt} \int_{\Omega_c} \phi \, d\mathbf{x} = \int_{\Omega_c} \frac{d\phi}{dt} \, d\mathbf{x} = \int_{\Omega_c} \frac{\partial \phi}{\partial t} \Big|_{\mathbf{x}^c} + \frac{\partial \phi}{\partial x_i^c} v_i^c \, d\mathbf{x} = \int_{\Omega_c} \frac{\partial \phi}{\partial t} \Big|_{\mathbf{x}^c} \, d\mathbf{x}$$



## 3. Field equations

For further reading see Reddy 5.1, 5.2.3, 5.3.1, 5.3.3., 5.4.1, 5.4.2.

### 3.1 Physical quantities of a continuum

We consider a body occupying a region  $\Omega$  at time  $t$ . The state of the body is assumed to be given by the quantities: mass  $M$ ; momentum  $\mathbf{P}$ ; angular momentum  $\mathbf{N}$ ; kinetic energy  $K$  and internal energy  $U$ . Before giving the expressions of these quantities we remind us the corresponding expressions of  $\mathbf{P}$ ,  $\mathbf{N}$  and  $K$  are given for a point mass  $m$  as:

$$\mathbf{P} = m \mathbf{v} \quad \text{or} \quad P_i = m v_i$$

$$\mathbf{N} = m \mathbf{x} \times \mathbf{v} \quad \text{or} \quad N_i = m e_{ijk} x_j v_k$$

$$K = m |\mathbf{v}|^2/2 \quad \text{or} \quad K = m v_i v_i/2$$

With this at hand and by assuming a density field  $\rho(\mathbf{x}, t)$  and a velocity field  $\mathbf{v}(\mathbf{x}, t)$  then the  $M$ ,  $\mathbf{P}$ ,  $\mathbf{N}$  and  $K$  for a body can be expressed as:

$$M = \int_{\Omega} \rho \, d\mathbf{x}, \quad (3.1)$$

$$P_i = \int_{\Omega} \rho v_i \, d\mathbf{x}, \quad (3.2)$$

$$N_i = \int_{\Omega} e_{ijk} x_j \rho v_k \, d\mathbf{x}, \quad (3.3)$$

$$K = \int_{\Omega} \frac{1}{2} \rho v_i v_i \, d\mathbf{x}. \quad (3.4)$$

In addition to these quantities the internal energy  $U$  is also introduced.  $U$  represents energy such as strain energy and thermal energy which together with the kinetic energy sums up to total energy of the body. Later  $U$  will be given more explicitly but at this stage we assume that the body has an internal energy density field  $e(\mathbf{x}, t)$  such that:

$$U = \int_{\Omega} \rho e \, d\mathbf{x}. \quad (3.5)$$

### 3.2 Input quantities

A schematic figure of a continuous body is given in figure 3.1 with the field variables  $\rho$ ,  $\mathbf{v}$  and  $e$ .

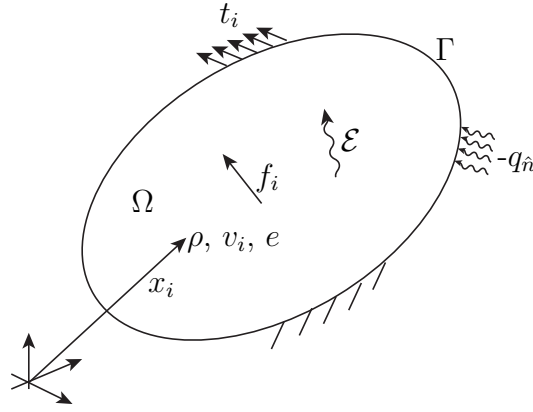


Figure 3.1: Illustration of a continuum  $\Omega$  with boundary  $\Gamma$ .

Now we assume that the body is subjected input quantities that can change the state of the body, see figure 3.1. The mechanical loading is given by a volume force  $\mathbf{f}$  (force per unit mass) and a boundary load  $\mathbf{t}$  (force per unit area). Thereby the total force  $\mathbf{F}$  and moment  $\mathbf{M}$  and mechanical power input to the body become:

$$F_i = \int_{\Omega} \rho f_i \, d\mathbf{x} + \oint_{\Gamma} t_i \, ds, \quad (3.6)$$

$$M_i = \int_{\Omega} e_{ijk} x_j \rho f_k \, d\mathbf{x} + \oint_{\Gamma} e_{ijk} x_j t_k \, ds., \quad (3.7)$$

$$\dot{W} = \int_{\Omega} \rho f_i v_i \, d\mathbf{x} + \oint_{\Gamma} v_i t_i \, ds, \quad (3.8)$$

Additionally, the body is subjected to the internal heat source  $\mathcal{E}$  (energy per unit mass) and heat input  $-q_{\hat{n}}$  (energy per unit area) resulting in the heat power input:

$$\dot{H} = \int_{\Omega} \rho \mathcal{E} \, d\mathbf{x} + \oint_{\Gamma} -q_{\hat{n}} \, ds \quad (3.9)$$

### 3.3 Physical conservation principles

Now the physical conservation principles are used to define how the state i.e. the mass  $M$ , the linear momentum  $\mathbf{P}$ , the angular momentum  $\mathbf{N}$  and the total energy  $K + U$  change of the body change with the mechanical and heat input.

#### 3.3.1 Conservation of mass

Mass is in classical mechanics assumed to be conserved which can be written as:

$$\dot{M} = \frac{d}{dt} \int_{\Omega} \rho \, d\mathbf{x} = 0 \quad (3.10)$$

By using Reynold's transport theorem (2.24) we obtain:

$$\dot{M} = \frac{d}{dt} \int_{\Omega} \rho \, d\mathbf{x} = \int_{\Omega} (\dot{\rho} + \rho v_{i,i}) \, d\mathbf{x} \quad (3.11)$$

The mass conservation is assumed for all choices of  $\Omega$  (this argumentation is called localization) whereby:

$$\dot{\rho} + \rho v_{i,i} = 0 \quad \text{in } \Omega \quad (3.12)$$

This equation is called the *continuity equation*.

The continuity equation can be used together with Reynold's transport theorem to show that for an arbitrary field quantity  $\phi$ :

$$\frac{d}{dt} \int_{\Omega} \rho \phi \, d\mathbf{x} = \int_{\Omega} \frac{d}{dt} (\rho \phi) + \rho \phi v_{i,i} \, d\mathbf{x} = \int_{\Omega} \rho \dot{\phi} \, d\mathbf{x} \quad (3.13)$$

This result is denoted the modified Reynold's transport theorem.<sup>1</sup>

**Problem 26** 5.6 a in Reddy

**5.6** Determine whether the following velocity fields for an incompressible flow satisfies the continuity equation:

$$(a) \quad v_2(x_1, x_2) = -\frac{x_1}{r^2}, \quad v_1(x_1, x_2) = -\frac{x_2}{r^2}, \quad \text{where } r^2 = x_1^2 + x_2^2.$$

*Answer:* The continuity equation is indeed fulfilled.

### 3.3.2 Conservation of linear and angular momentum - Newton's laws

Newton's laws state that the material time derivative of the linear momentum  $\mathbf{P}$  and the angular momentum  $\mathbf{N}$  are determined by the applied force  $\mathbf{F}$  and moment  $\mathbf{M}$  as follows:

$$\begin{aligned} \dot{P}_i &= F_i \\ \dot{N}_i &= M_i \end{aligned}$$

By using modified Reynold's transport theorem (3.13) and equations (3.2) as well as (3.6) we obtain:

$$\dot{P}_i = \int_{\Omega} \rho \dot{v}_i \, d\mathbf{x} = \int_{\Omega} \rho f_i \, d\mathbf{x} + \oint_{\Gamma} t_i \, ds \quad (3.14)$$

The next step is to use Cauchy's stress principle  $t_i = \sigma_{ji} \hat{n}_j$  and the divergence theorem:

$$\int_{\Omega} \rho \dot{v}_i - \sigma_{ji,j} - \rho f_i \, d\mathbf{x} = 0 \quad (3.15)$$

The localization argument now yields the momentum equation:

$$\sigma_{ji,j} + \rho f_i = \rho \dot{v}_i \quad (3.16)$$

<sup>1</sup>It can also be shown by instead integrating over the mass:  $\frac{d}{dt} \int_{\Omega} \rho \phi \, d\mathbf{x} = \frac{d}{dt} \int_m \phi \, dm = \int_m \dot{\phi} \, dm = \int_{\Omega} \rho \dot{\phi} \, d\mathbf{x}$

By using the same steps for the angular momentum  $\dot{N}_i = M_i$  we will arrive at the result that the stress tensor must be symmetric:

$$\boldsymbol{\sigma}^T = \boldsymbol{\sigma} \quad \text{or} \quad \sigma_{ij} = \sigma_{ji} \quad (3.17)$$

However, we leave those derivations to Hand-in assignment 12.

**Problem 27** 5.18 in Reddy

**5.18** If the stress field in a body has the following components in a rectangular Cartesian coordinate system

$$[\boldsymbol{\sigma}] = a \begin{bmatrix} x_1^2 x_2 & (b^2 - x_2^2)x_1 & 0 \\ (b^2 - x_2^2)x_1 & \frac{1}{3}(x_2^2 - 3b^2)x_2 & 0 \\ 0 & 0 & 2bx_3^2 \end{bmatrix},$$

where  $a$  and  $b$  are constants, determine the body force components necessary for the body to be in equilibrium.

*Answer:*

$$f_1 = 0 \quad f_2 = 0 \quad f_3 = -\frac{4abx_3}{\rho}$$

**Problem 28** 5.12 in Reddy

**5.12** Use the continuity (i.e., conservation of mass) equation and the equation of motion to obtain the so-called conservation form of the momentum equation

$$\frac{\partial}{\partial t} (\rho \mathbf{v}) + \text{div} (\rho \mathbf{v} \mathbf{v} - \boldsymbol{\sigma}) = \rho \mathbf{f}.$$

**Hand-in assignment 11** A rectangular disc with width  $b_1$ , height  $b_2$  and thickness  $h$  rotates with angular velocity  $\boldsymbol{\omega} = \omega_0 e^{-t}$  around  $x_3$  axis.

Assume that the density is constant  $\rho_0$ .



Determine the angular momentum of the disc.

**Hand-in assignment 12** Use the principle of angular momentum to show that the stress tensor  $\boldsymbol{\sigma}$  is symmetric. Hint: see Reddy 5.3.3.

### 3.3.3 Conservation of energy - 1st law of thermodynamics

The 1st law of thermodynamics says that the material time derivative of the total energy of a body is equal to the power input:

$$\dot{K} + \dot{U} = \dot{W} + \dot{H} \quad (3.18)$$

If we now use modified Reynold's transport theorem (3.8) and equations (3.4), (3.5), (3.8) as well as (3.9) we obtain:

$$\int_{\Omega} \rho v_i \dot{v}_i + \rho \dot{e} \, d\mathbf{x} = \int_{\Omega} \rho f_i v_i \, d\mathbf{x} + \oint_{\Gamma} t_i v_i \, ds + \int_{\Omega} \rho \mathcal{E} \, d\mathbf{x} + \oint_{\Gamma} -q_{\hat{n}} \, ds \quad (3.19)$$

Before we can use the localization argument the boundary integrals must be changed to volume integrals. The first of the boundary integrals is re-written using the Cauchy's stress theorem as follows:

$$\oint_{\Gamma} t_i v_i \, ds = \oint_{\Gamma} \hat{n}_j \sigma_{ji} v_i \, d\mathbf{x} = \int_{\Omega} \sigma_{ji,j} v_i + \sigma_{ji} v_{i,j} \, d\mathbf{x} \quad (3.20)$$

The second boundary integral is re-written by assuming that the heat flux  $q_{\hat{n}}$  is given by the heat flux vector  $\mathbf{q}$  according to:

$$q_{\hat{n}} = \hat{\mathbf{n}} \cdot \mathbf{q} = \hat{n}_i q_i \quad (3.21)$$

thereby the divergence theorem gives us:

$$\oint_{\Gamma} -q_{\hat{n}} \, ds = \int_{\Omega} -q_{i,i} \, d\mathbf{x} \quad (3.22)$$

Now the 1st law of thermodynamics can be written as:

$$\int_{\Omega} \rho v_i \dot{v}_i + \rho \dot{e} - \rho f_i v_i - \sigma_{ji,j} v_i - \sigma_{ji} v_{i,j} - \rho \mathcal{E} + q_{i,i} \, d\mathbf{x} = 0 \quad (3.23)$$

The balance of linear momentum (3.16) now together with the localization argument results in the energy equation:

$$\rho \dot{e} - v_{i,j} \sigma_{ij} + q_{i,i} = \rho \mathcal{E}. \quad (3.24)$$

**Hand-in assignment 13** By taking the material time derivative of the kinetic energy in (3.4) and using the balance of linear momentum show that for a purely mechanical problem ( $\dot{H} = 0$ ):

$$\dot{K} + \int_{\Omega} \sigma_{ij} v_{i,j} d\mathbf{x} = \dot{W}$$

### 3.4 Summary of field equations and field variables

A summary of the field equations and the field variables for a continuum are shown in the table below.

Balance law	Field variables	No equations
$\dot{\rho} + \rho v_{i,i} = 0$	$\rho, v_i$	1
$\rho \dot{v}_i = \sigma_{ji,j} + \rho f_i$	$\sigma_{ij}$	3
$\sigma_{ij} = \sigma_{ji}$	$\sigma_{ij}$	3
$\rho \dot{e} = \sigma_{ij} v_{i,j} - q_{i,i} + \rho \mathcal{E}$	$e, q_i$	1
	Tot. 17	Tot. 8

By counting the number of equations and unknowns we can conclude that 9 additional equations that must be formulated. These equations are the constitutive models that should mimic the material behaviour observed in experiments.

## 4. Constitutive models

The standard way to define constitutive models in solid mechanics, fluid mechanics and heat transfer is to describe how internal energy  $e$ , stress  $\boldsymbol{\sigma}$  and heat flux  $\mathbf{q}$  depend on other field variables such as density  $\rho$ , temperature  $\theta$ , temperature gradient  $\theta_{,i}$ , displacement gradient  $u_{i,j}$  and velocity gradient  $v_{i,j}$ . These models with their material parameters are based on experimental observations. In this section we will merely introduce some of most common (and simplest) constitutive models for heat transfer, fluids and solids.

The constitutive models are determined by the material properties. The material can be homogeneous meaning that properties are the same in the body  $\Omega$  otherwise the material is heterogeneous. If the properties are the same in all directions then the material is called isotropic. For some materials the properties are anisotropic. Examples of the latter are: wood, composites and fibre reinforced concrete.

### 4.1 Fourier's law of thermal conductivity

For further reading see Reddy 6.4.2.

Heat can be transferred by convection (motion of fluid), radiation (electromagnetics) and conduction (diffusion processes). For heat conduction the standard constitutive model is Fourier's law. For an isotropic material this law takes the form:

$$q_i = -k \Theta_{,i} \quad (4.1)$$

where the linear coefficient  $k$  is the thermal conductivity and  $\Theta$  is the temperature. The temperature  $\Theta$  is now assumed to give the internal energy  $e$  according to:

$$e = c_p \Theta \quad (4.2)$$

where the constant  $c_p$  is the heat capacity of the material. If we consider a purely thermal problem (i.e. assuming  $\sigma_{ij} = 0$ ) then the energy equation (3.24) reads as follows:

$$\rho \dot{e} = -q_{i,i} + \rho \mathcal{E}$$

By inserting (4.1) and (4.2) then we obtain the transient heat conduction equation:

$$\rho c_p \dot{\Theta} = k \Theta_{,ii} + \rho \mathcal{E} \quad (4.3)$$

## 4.2 Viscous fluids

For further reading see Reddy 6.3.3.

The simplest possible constitutive of a fluid is an ideal fluid. In this model the stress is assumed to be purely volumetric:

$$\sigma_{ij} = -p(\rho, \Theta) \delta_{ij} \quad (4.4)$$

This means that such a fluid cannot sustain shear stresses. The pressure  $p$  is assumed to follow the ideal gas law

$$p(\rho, \Theta) = \rho R \Theta / m_g \quad (4.5)$$

where  $R$  is the gas constant,  $m_g$  is the mean molecular mass of the gas.

Most fluids are not "ideal" since they are a bit "sticky" and are able to sustain shear stresses. Therefore, a viscous stress  $\boldsymbol{\tau}$  is introduced for and the stress  $\boldsymbol{\sigma}$  is additively decomposed according to

$$\sigma_{ij} = -p(\rho, \Theta) \delta_{ij} + \tau_{ij} \quad (4.6)$$

For the case of an isotropic *Newtonian viscous fluid* we assume that the viscous stress  $\boldsymbol{\tau}$  is linear in terms of the strain rate tensor  $\mathbf{D}$  as follows

$$\tau_{ij} = \lambda^* D_{kk} \delta_{ij} + 2\mu^* D_{ij} \quad (4.7)$$

where the strain rate tensor  $\mathbf{D}$  is defined from as the symmetric part of the velocity gradient:

$$D_{ij} = \frac{1}{2} (v_{i,j} + v_{j,i}). \quad (4.8)$$

In (4.7) the material parameters  $\lambda^*$  and the dynamic (shear) viscosity  $\mu^*$  were introduced. Now the stress becomes:

$$\sigma_{ij} = -p(\rho, \Theta) \delta_{ij} + \lambda^* \delta_{ij} D_{kk} + 2\mu^* D_{ij} \quad (4.9)$$

The mechanical pressure  $p_{\text{mech}} = -\sigma_{mm}/3$  can be computed as:

$$p_{\text{mech}} = -\frac{1}{3} \sigma_{mm} = p(\rho, \Theta) - \left( \lambda^* + \frac{2}{3} \mu^* \right) D_{kk} \quad (4.10)$$

If we introduce the Stoke's condition that  $p_{\text{mech}} = p(\rho, \Theta)$  then we can for a Newtonian viscous fluid obtain that

$$\sigma_{ij} = \sigma'_{ij} + \frac{\sigma_{kk}}{3} \delta_{ij} = 2\mu^* D'_{ij} - p(\rho, \Theta) \delta_{ij} \quad (4.11)$$



where  $\boldsymbol{\sigma}'$  and  $\boldsymbol{D}'$  are the deviatoric stress and deviatoric strain rate tensor, respectively. Stoke's condition means that the pressure in the fluid is strain rate independent. The Navier-Stoke's equations are now simply obtained by inserting (4.11) into balance of linear momentum (3.16) together with the continuity equation.

Often for fluids one can assume that they are incompressible. From the conservation of mass and the incompressibility  $\dot{\rho} = 0$  it follows from the continuity equation (3.12) that:

$$v_{i,i} = D_{ii} = 0$$

In this case  $\boldsymbol{D}' = \boldsymbol{D}$  and, without using Stoke's condition, we obtain (4.11) from (4.9).

**Problem 29** Assume the constitutive equation  $\sigma_{ij} = (-p + \lambda D_{kk}) \delta_{ij} + 2\mu D_{ij}$ . Show that the equations of motion can be expressed in the velocity field as:

$$\rho \dot{v}_i = \rho f_i - p_{,i} + (\lambda + \mu) v_{j,ij} + \mu v_{i,jj}.$$

### 4.3 Linear elastic isotropic solids

The constitutive model for linear elasticity is denoted Hooke's law. Originally the law was defined for a linear spring but generalized to an isotropic solid it reads as follows

$$\sigma_{ij} = \lambda \epsilon_{kk} \delta_{ij} + 2\mu \epsilon_{ij} \quad (4.12)$$

where  $\epsilon_{ij}$  is the small strain tensor defined in (2.16). A small strain assumption has been made and it is therefore the small strain tensor can be used. This also means that the density  $\rho$  can be assumed to be approximately constant. The model parameters  $\lambda$  and  $\mu$  are the Lamé's constants, which are related to Young's modulus  $E$  and Poisson's ratio  $\nu$  as follows

$$\lambda = \frac{E\nu}{(1+\nu)(1-2\nu)}, \quad \mu = \frac{E}{2(1+\nu)}$$

**Hand-in assignment 14** The equations

$$\mu u_{i,jj} + (\lambda + \mu) u_{j,ji} + \rho f_i = \rho \ddot{u}_i.$$

are called Navier's equations and may be used to solve elastodynamic problems with displacement-type boundary conditions. Derive these equations by combining the momentum equation and Hooke's law.